

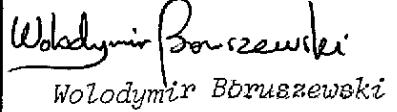
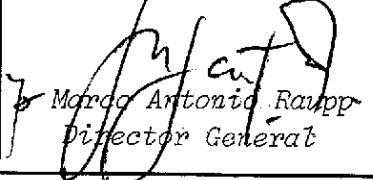
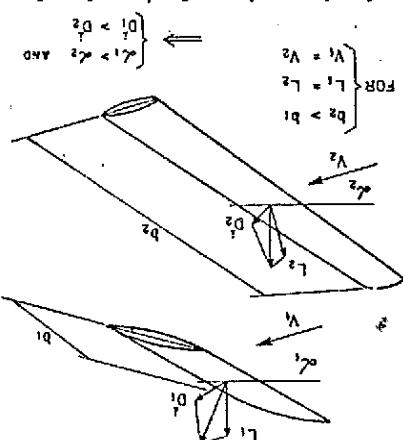
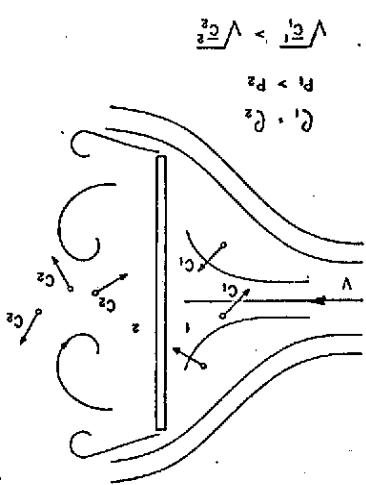
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14. Abstract/Notes			
<p>The application of gas dynamics concepts can clarify and explain many aspects of aerodynamic phenomena. In the case of friction drag, this approach suggests some new possibilities for its reduction, such as the use of "ultrasmooth" surfaces having specific physical and chemical characteristics or used in conjunction with a noble gas lubricating layer. Such possibilities are discussed in the text which, due to the multidisciplinary knowledge involved, is kept at a conceptual and heuristic level.</p>			
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Fig. 1 - Drag as molecular collision effects.

At same speed, the number of air molecules colliding be deflected with a greater angle is lower (wing area role is negligible). To result in the same lift they must with the shorter span wing is lower (wing area role is greater).



In incompressible flow the number of molecules in front of and behind the plate are the same. Molecular speed being function only of pressure and temperature, collisions in front of plate occur with higher speeds.



A discrete approach of pressure and induced drag. In the study of gas motion phenomena has evolved in distance by trailing vortices, like gravity, magnetic or other field forces; in Figure 1 we attempt to show

The application of gas dynamics concepts in clarifying and explaining many aspects of aerodynamic phenomena. In the case of friction drag, this approach suggests some new possibilities for its reduction, such as the use of "ultrasmooth" surfaces having specific physical and chemical characteristics or used in conjunction with a notable gas lubricating layer. Such possibilities are discussed in the text which, due to the multidisciplinary knowledge involved, is kept at a conceptual and heuristic level.

ABSTRACT



The book gives a false physical idea of which many aerodynamic phenomena is applicable to life, pressure etc. This is due to the fact that fluid and solid molecules colliding at body surface. All aerodynamic forces acting on a body moving in a real continuum must be present in every aerodynamic gas continuum must be present in real continuum hypothesis assumed for the simplest calculations of continuum bulk properties. For simplified cases such as creeping flows, continuum hypotheses fails, such as in rarefied flows, statistical formulation is reserved to cases where the continuum hypothesis is cumulative So, the discrete approach with its cumbersome phenomena, most of them stationary. $\lambda/C = 10-10^6$ s. and the time scale of the aerodynamic over simplification. However this is fully justified if we consider the time between molecular collisions, space" a continuum seems, at a first glance, an present larger diameters, to consider this "empty Although oxygen and carbon dioxide molecules mean collision free distance: (λ) ~ 600 m/s mean quadratic molecular speed V^2 ~ 300 m/s molecular diameter 4 μ we have at S.L. pressure and 15°C [1]: Let us look, for instance, at the characteristics of nitrogen, one of main air constituents, for which continuum", since the real air is far from being approached, found without considering the discrete not to be found "why" of any aerodynamic phenomena seems technological success, anyway, the explanation of the gas modelling seems to have hampered an equivalent evolution, with technological solutions going from builtins to vehicles, these ranging from ultralights continuum assumption adopted has permitted an enormous in the first, the simplest solution resulting from the two distinct sciences: aerodynamics and gas dynamics. In the study of gas motion phenomena has evolved in distance by trailing vortices, like gravity, magnetic or other field forces; in Figure 1 we attempt to show

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INTRODUCTION

Concerning friction drag forces, we intend to examine it a little further, in an effort to find new possibilities for its reduction, but when trying to do so, we will be confronted with subjects dealing with solid state and quantum physics, surface science, etc., all outside the aerodynamic and fluid mechanical engineer's as well as the author's experience.

As a consequence, this note was kept only at a conceptual level and was written with the main objective of bringing the attention of the specialists of these fields to points of common interest with aerodynamics.

THE FRICTION DRAG AND THE "NO SLIP CONDITION"

In aerodynamics, the flow generated forces tangential to body surface are called friction forces and their resultant in the flow direction is called friction or surface drag.

For streamlined bodies such as airplanes and gliders, nearly all friction forces act as drag and can represent the major part of the total drag for a cruising airplane or a high speed flying glider, which is not the case for blunt bodies or satellites.

Using continuum fluid mechanics, the friction forces can be computed for various flow regimes and types using the highly developed "Boundary Layer Theory" [2] and the basic relation:

$$\tau = \mu dV/dy ,$$

where τ is the tangential force shear stress and dV/dy is the continuum fluid velocity gradient at the surface normal direction, both computed at the body surface, μ is a fluid property, the viscosity, which leads to an alternative name to the friction drag, that is, viscous drag, and to the corollary assumption that viscosity is the only responsible to this drag!

In reality, when doing so, we take for granted one of the basic hypothesis of the boundary layer theory: the "no slip condition" which states that the continuum tangential flow velocity vanishes at the solid surface, that is, $V \rightarrow 0$ when $y \rightarrow 0$.

This no slip condition has been extensively confirmed by tests, holding not only in flows where the "Knudsen number", which relates a flow characteristic length to the molecular collision free path, approaches the unit ($\lambda/L \rightarrow 1$).

Typical examples are rarefied gas flows and conditions at the very beginning of the boundary layer.

Although fluid dynamics text books quote this no slip condition "en passant" without further explanations, we must recognise its important role to the friction drag onset and that the fluid viscosity role is only to determine how this zero velocity propagates into the boundary layer.

As already said, the momentum exchange which resulted in the friction drag has already occurred at surface in fluid to solid molecular collisions!

All this may be more evident when we consider, for instance the events in a boundary layer of a surface starting from rest.

Since, for usual flows, pressure and, less frequently, temperature are constant across the boundary layer, the surface colliding molecules and the outside boundary layer "free flow" molecules have nearly the same speed, it is intriguing to know why the continuum tangential velocities always vanish even for very smooth surfaces.

Prandtl, who established the boundary layer theory basis, considering experimental effects of roughness on drag, used to ponder that solid surface contained more voids than matter and that fluid molecules, after penetrating these voids, returned to surface having lost all tangential momentum.

Today we know that, although fluid diffusion into solid may occur, solid molecular distances are not so great to allow a generalised penetration, except in pores of macromolecular dimensions.

What seems to happen is that even for a perfectly smooth surface such as that of a single crystal, fluid molecular tangential velocities are brought to zero value by molecular attraction forces known as "Van der Waals" forces.

These forces govern the physical adsorption phenomena and in conjunction with the chemisorption forces are fundamental in technological processes such as catalysis, bonding, thin film formation, etc.

Although they are considered "weak" forces when compared to other atomic or molecular forces (see table 1), they result from dipole attractions present in polar fluid molecules or induced in nonpolar ones and have sufficient strength to bring slip (tangential) fluid molecular velocities to zero.

TABLE 1 - Order of Magnitude of Atomic and molecular forces

ATOMIC FORCES

NUCLEAR	2000 KeV
NUCLEUS/INTERNAL ELECTRON	20 KeV
NUCLEUS(EXTERNAL ELECTRON)	10 KeV

MOLECULAR FORCES

ION OR "METALLIC" BOND	4 eV
VAN DER WAALS BOND	1-7 eV
HYDROGEN	0,5 eV

obs.: 1 eV = 23 Kcal/mol.

The complete model of fluid and solid substrate molecular collision processes is still to be fully developed for liquids or gases, and according to Ref. [3] quantum mechanic models are a must to account for the molecular electron clouds interactions occurring in the collision process.

In most cases adsorption and sometimes chemical reactions may be present so that for practical flows a theoretical complete picture of the no slip condition and any action to modify it seem to be far out of reach of theoretical treatment (See Figure 2).

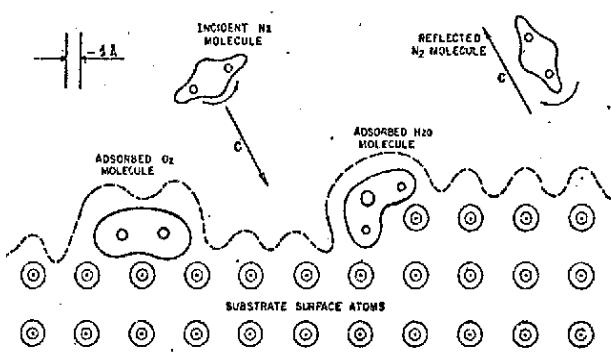
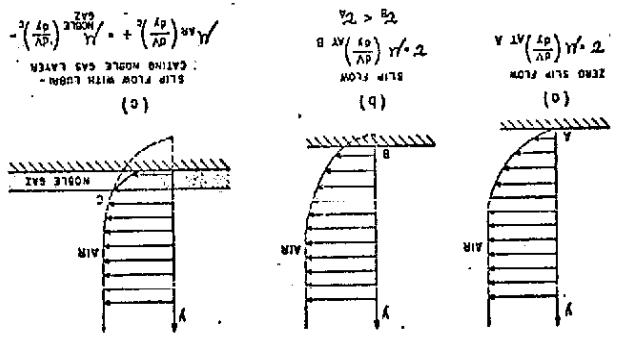


Fig. 2 - Schematic view of single crystal surface and air molecules interactions.

REFERENCE

- [1] Viscosity, W.G., & Krueger, Jr., C.H., "Introduction to Physical Gas Dynamics" Robert E. Krueger Publishing Co., 1965.
- [2] Schlichting, H., "Boundary Layer Theory" - MacGraw Hill Int. Book.
- [3] Lewis, B., and Anderson, J.C., "Nucleation and Growth of Thin Films", Academic Press, 1978.
- [4] Goodman, P.O., and Macmillan, H.V., "Dynamic of gas surface scattering" Academic Press, 1976.
- [5] Patereson, G.N., "Molecular Nature of Aerodynamics UTAS Publication, 1981.

Fig. 4 - Boundary Layer Profiles for Three Flow Types.



In addition, such surfaces will not need the protection means needed for other drag reducing devices such as boundary layer suction, being present only considered.

After all, neither insects nor dust will adhere where the air itself will refuse to do so.

Fig. 4 - Boundary Layer Profiles for Three Flow Types.

$$\frac{dy}{dx} = \frac{U_{\infty}(x)}{U_{\infty}} + \frac{1}{\lambda} \left(\frac{dy}{dx} \right)^2$$

(a)

$$\frac{dy}{dx} = \frac{U_{\infty}(x)}{U_{\infty}} + \frac{1}{\lambda} \left(\frac{dy}{dx} \right)^2$$

(b)

$$\frac{dy}{dx} = \frac{U_{\infty}(x)}{U_{\infty}} + \frac{1}{\lambda} \left(\frac{dy}{dx} \right)^2$$

(c)



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