



MINISTÉRIO DA CIÊNCIA E TECNOLOGIA  
INSTITUTO NACIONAL DE PESQUISAS ESPACIAIS

## *Experimental Tests of Thermal Contact Resistance between Satellite Electronic Boxes and Honeycomb Panels under Typical Flight Conditions*

**Rafael Lopes Costa, Valeri Vlassov**

[rafael.costa@inpe.br](mailto:rafael.costa@inpe.br), [valeri.vlassov@inpe.br](mailto:valeri.vlassov@inpe.br)

**Brazilian National Institute for Space Research (INPE), São José dos Campos-SP, Brazil**

# Introduction

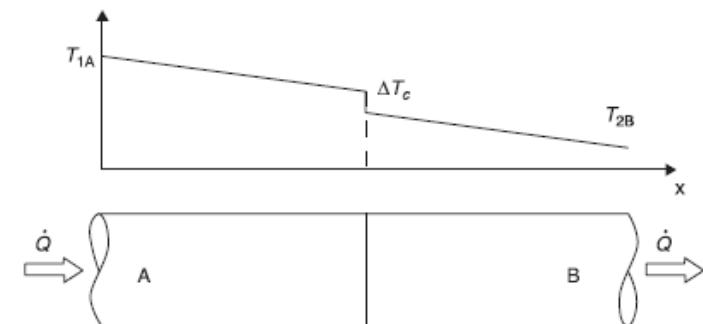
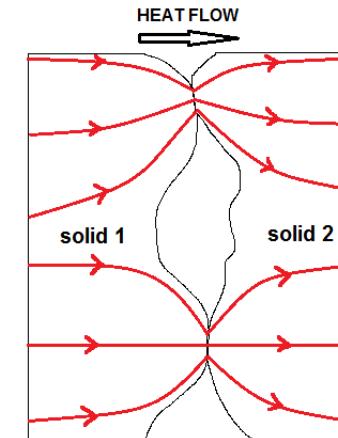
- When solids are placed into contact a **Thermal resistance** is created
- Effective contact area < Total interface area

- **Parameters** affecting the thermal contact resistance:

- Pressure
- Surface finishing
- Mechanical properties
- Presence of interface material

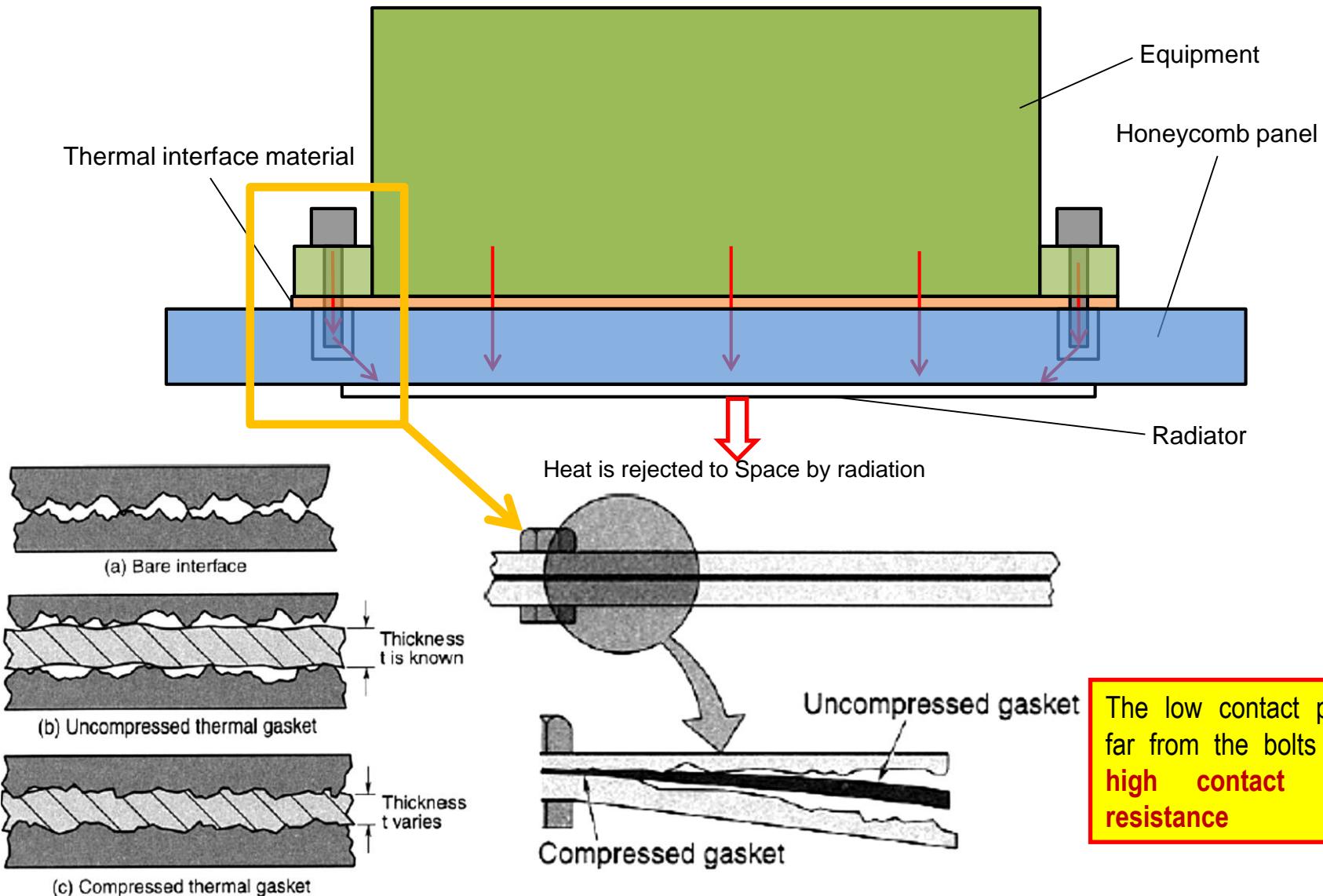


The spacecraft is under **VACUUM** in Space



# Introduction

## Satellite equipment/panel typical interface



# Study objectives

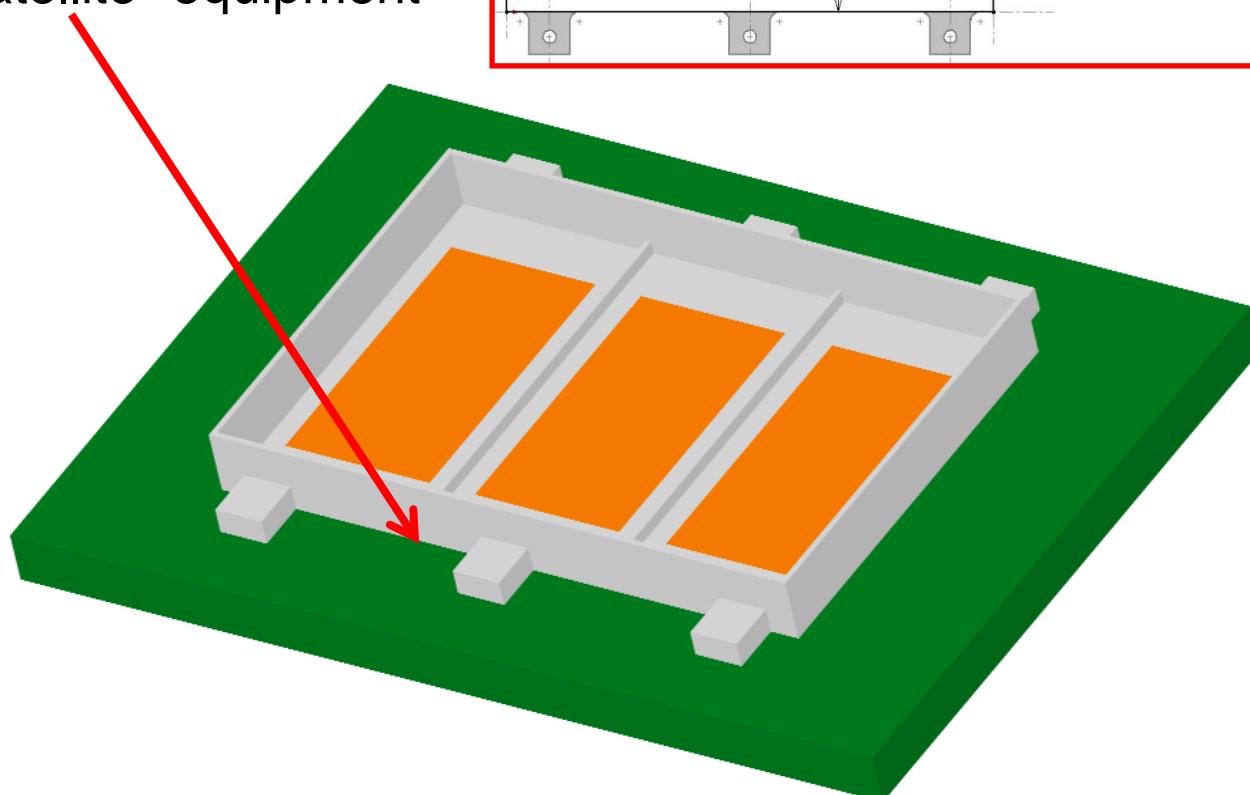
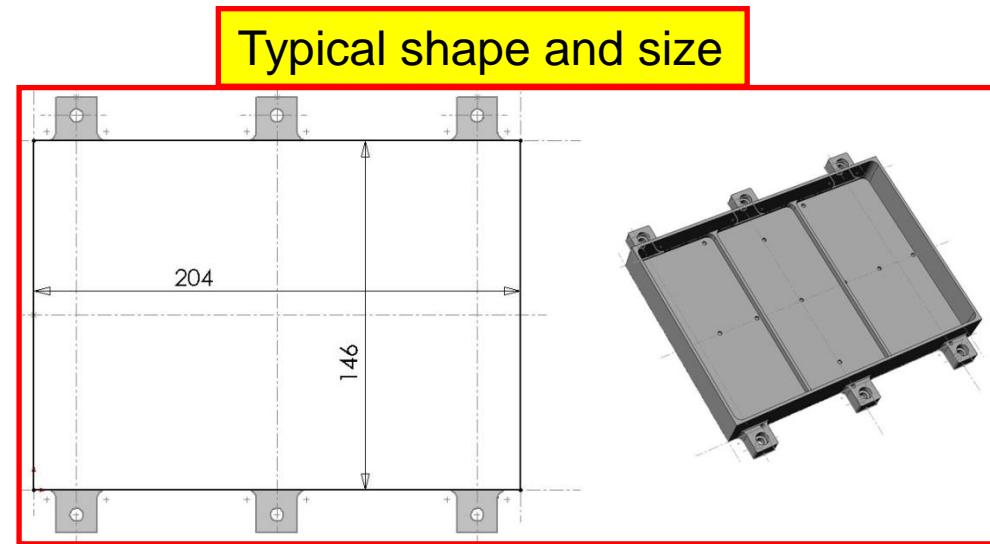
- a) Development of a **new test setup** that simulates the thermal interface between satellite equipment baseplates and honeycomb panels and allows multiple sets to be tested simultaneously under vacuum;
- b) Experimental **distribution of thermal resistance** of the interface with different materials;
- c) Experimental thermal contact resistance as a **function of temperature**;
- d) Impact of **flatness and roughness** of the surfaces on the thermal performance;

# Study objectives

- e) Study of thermal performance stability after temperature cycles;
- f) Influence of the non-homogeneous heat dissipation on the equipment;
- g) Identification of effective thermal resistance of the interface to be used in simplified TMMs;
- h) Comparison of thermal performance: Vacuum vs  $P_{amb}$

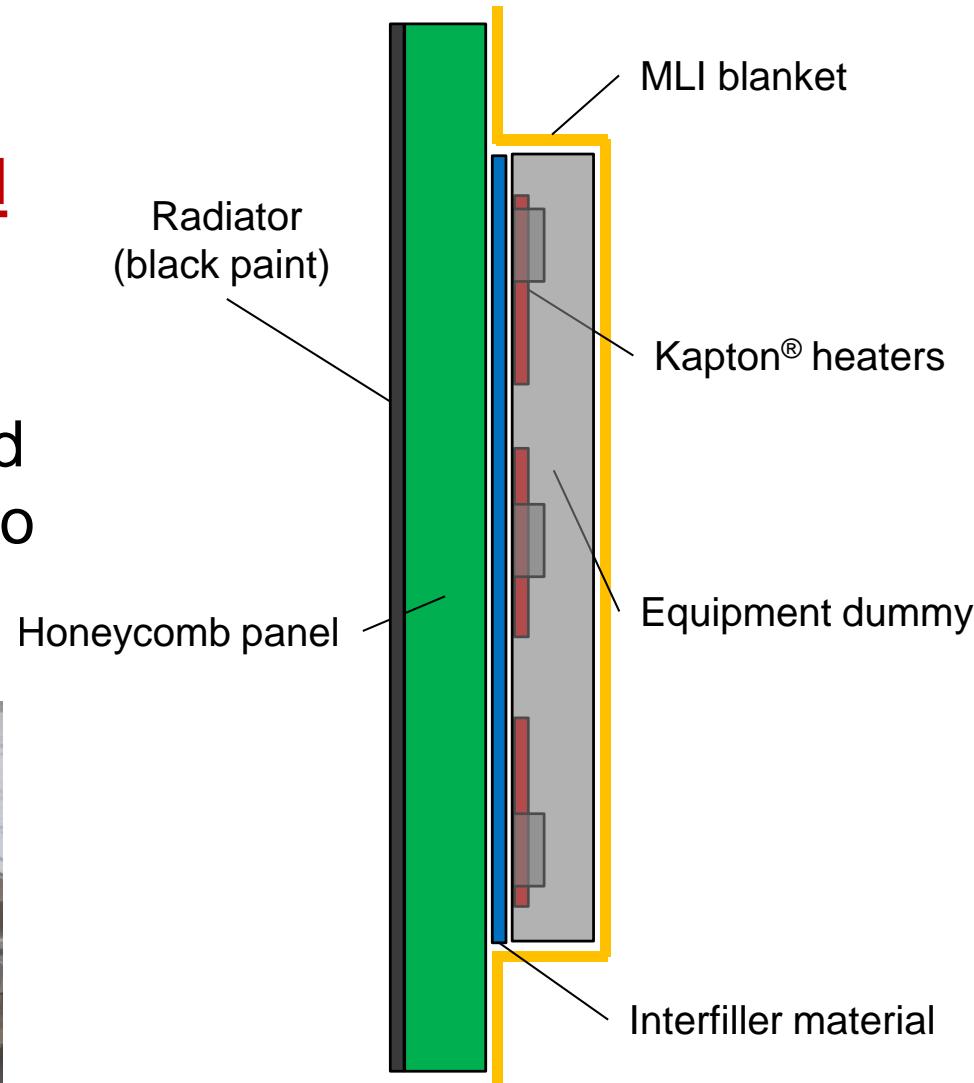
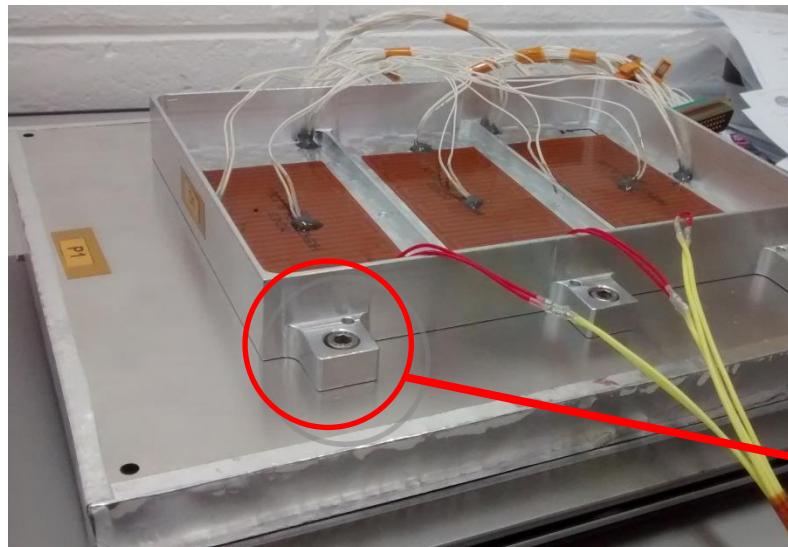
# Test setup

- Al Dummy for equipment baseplate simulation
- Piece of Al sandwich honeycomb panel
- Typical satellite equipment interface



# Test setup

- Honeycomb panel **painted in black** to perform like a radiator
- Equipment dummy covered with **MLI blanket** in order to insulate

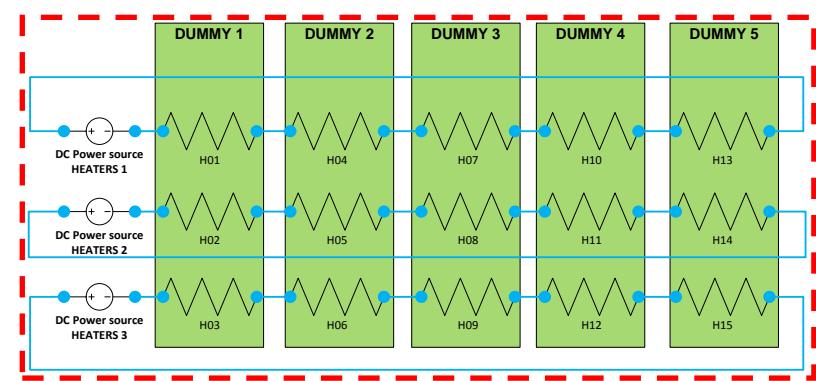
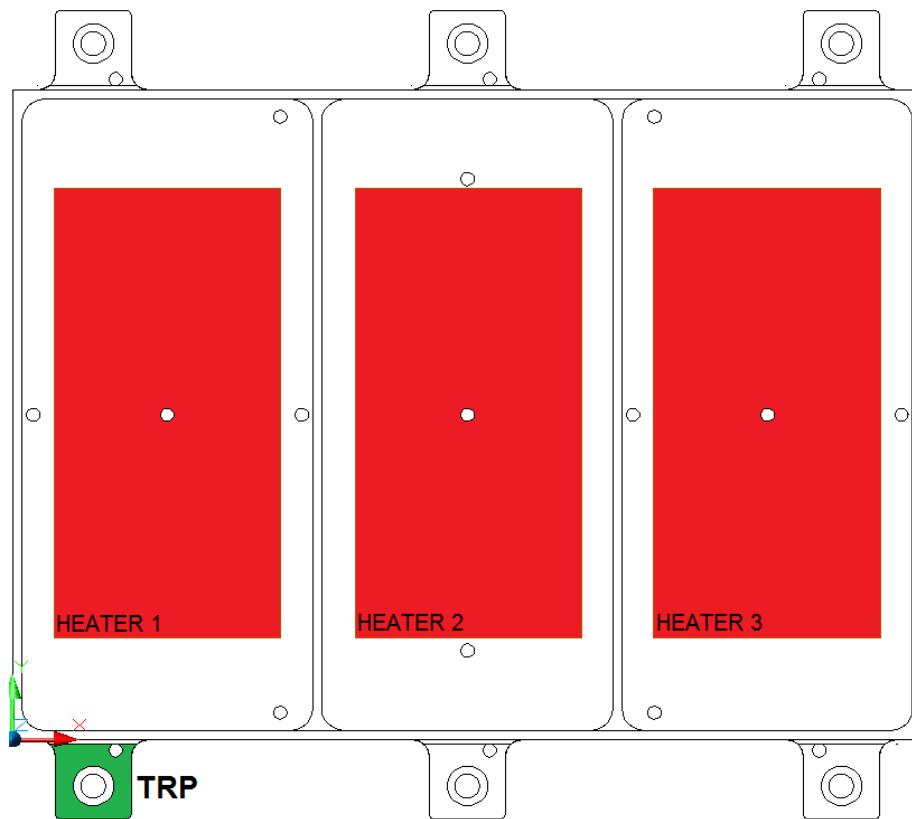


Torque according to the DCS documents  
5 Nm for M5 bolts

# Test setup

- Kapton patch heaters to simulate the equipment components heat dissipation
- Heat flux oriented towards the radiator

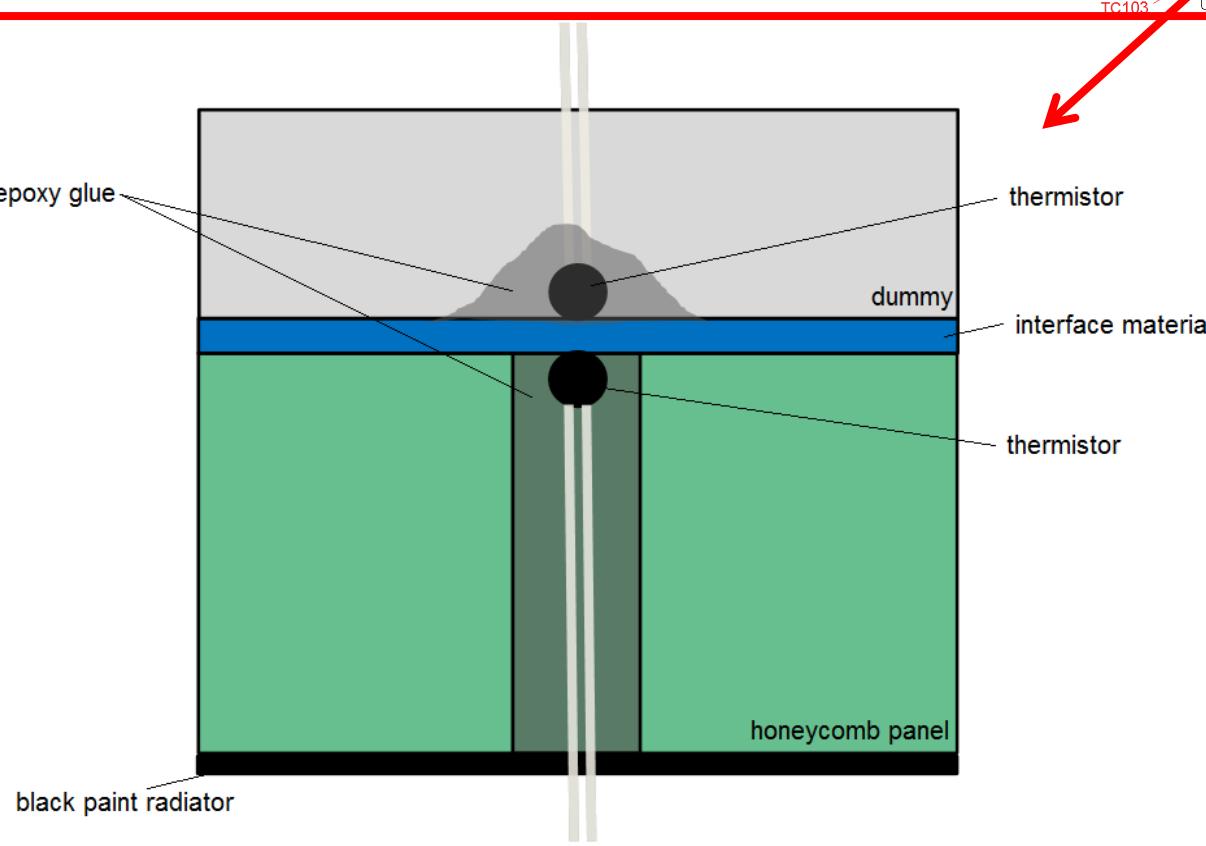
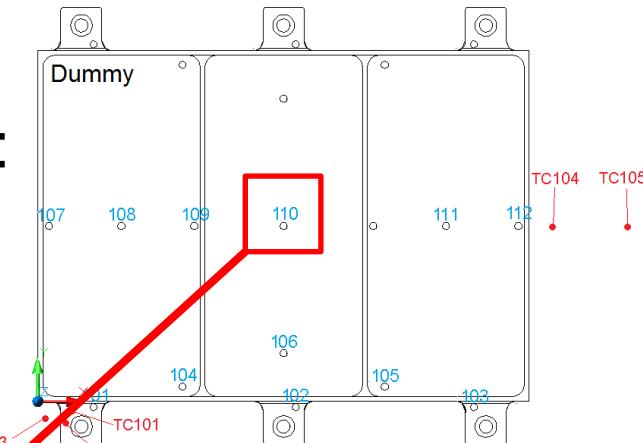
→ Through the interface



Operation mode	Description
H1+H2+H3	All heaters ON
H1	Only H1 is ON (close to the TRP)
H3	Only H3 is ON (far from the TRP)

# Test setup - sensors

Top view from the dummy

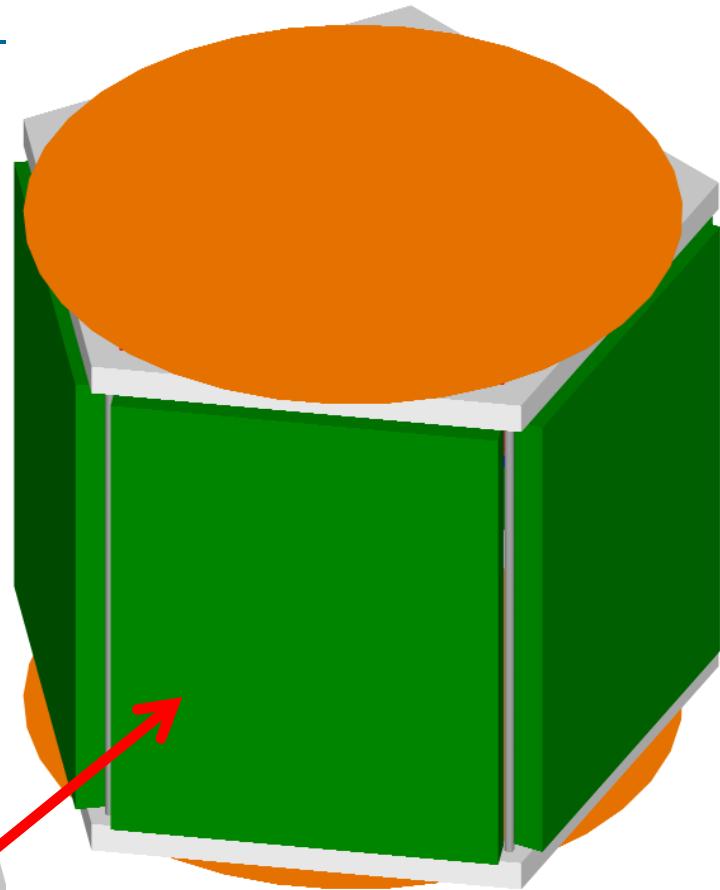
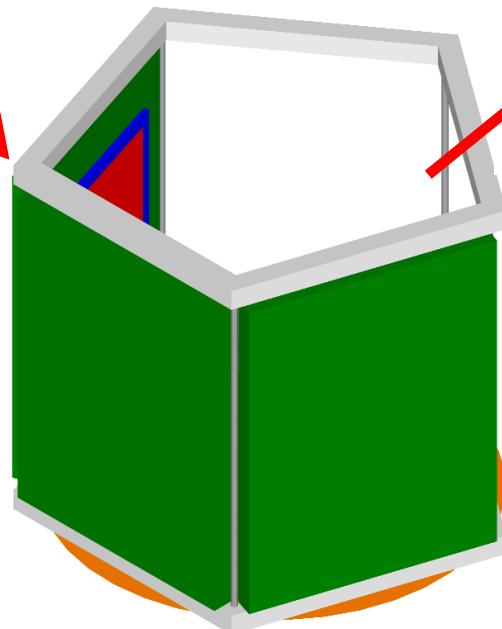
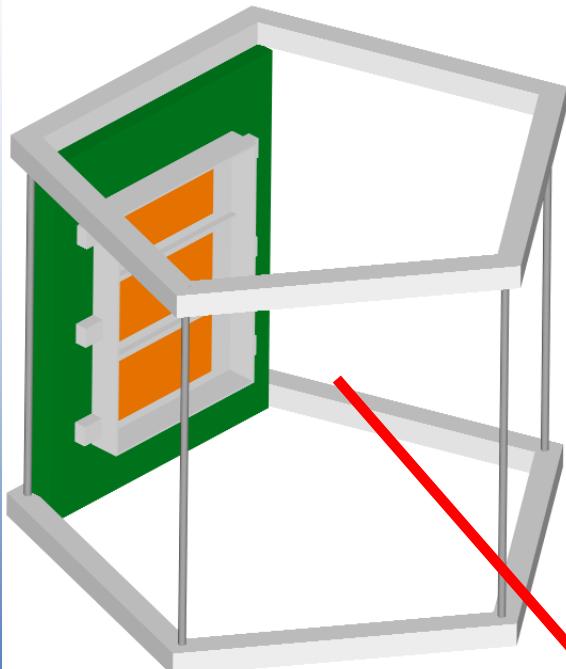


- 12 TMs on the dummy
- 12 TMs on the panel
- Corresponding TMs (in pairs)
- Individually calibrated TMs

Steinhart-Hart equation

$$T = -273.15 + \frac{1}{C1 + C2 \ln(R) + C3 \ln(R)^3}$$

# Test setup - assembly



- Test of 5 different thermal interfaces simultaneously
- Reusable apparatus

All interfaces are submitted to equivalent thermal conditions!

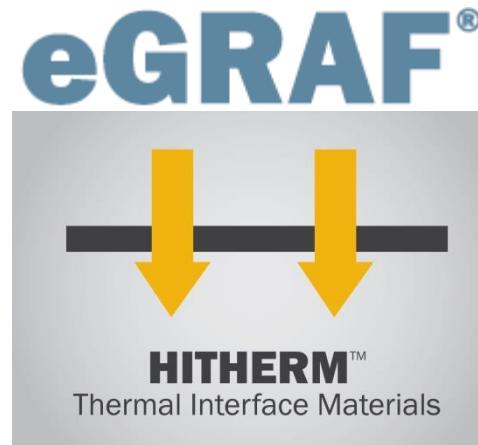
# Tested materials

Material	Manufacturer	Thickness (mm)	Thermal conductivity (W/m°C)
Bare (no material)	-	-	-
eGraf® Hitherm™ 1200 Series	Graftech	0.25	10.0 / 150.0
Indium foil (99.995%)	SMC (Shanghai Metal Corporation)	0.20	~90.0
Thermal Grease (340 HS Compound)	Dow Corning	0.20*	~2.0
RTV566 (cured-in)	Momentive	0.20*	0.3

\*at installation moment

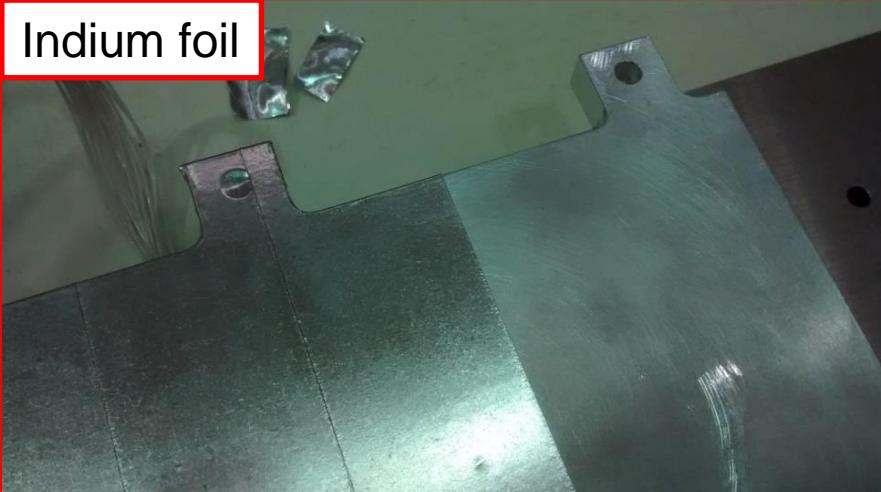


**MOMENTIVE™**  
inventing possibilities

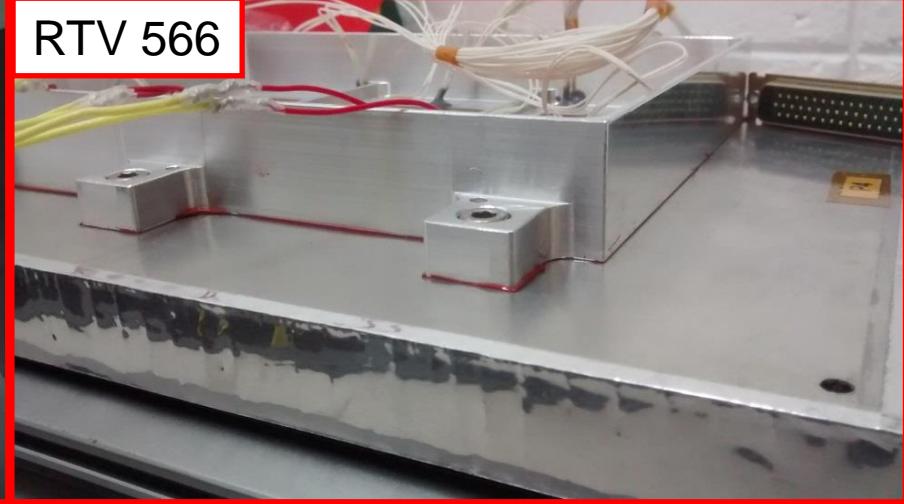


# Apparatus photos

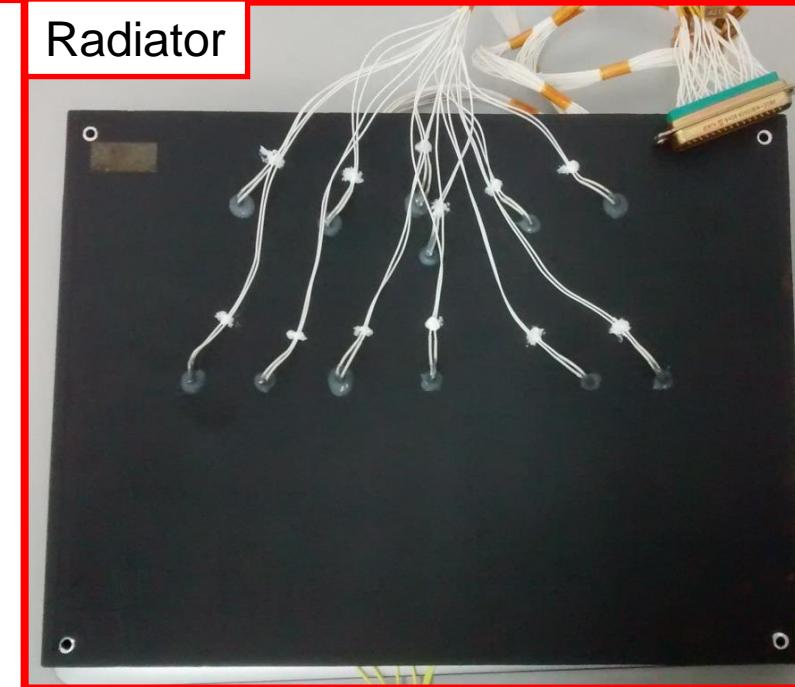
Indium foil



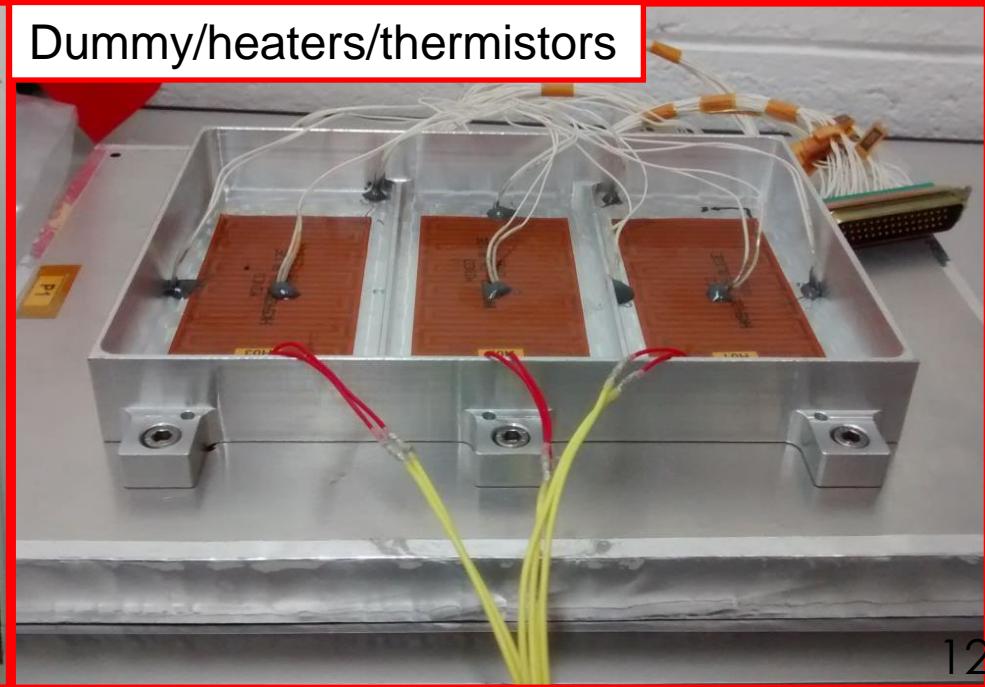
RTV 566



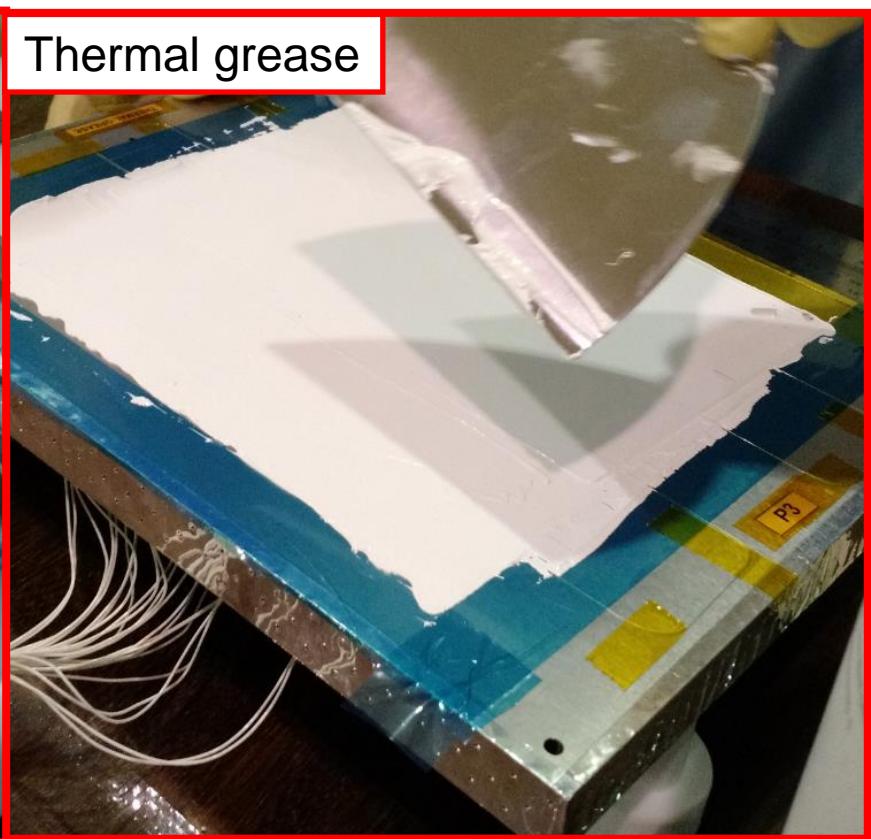
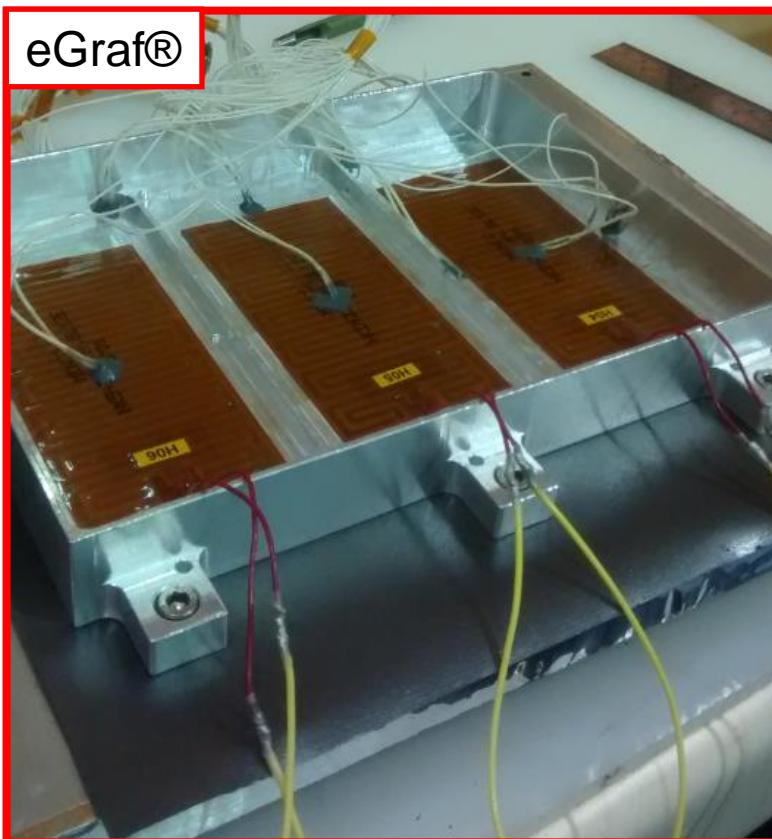
Radiator



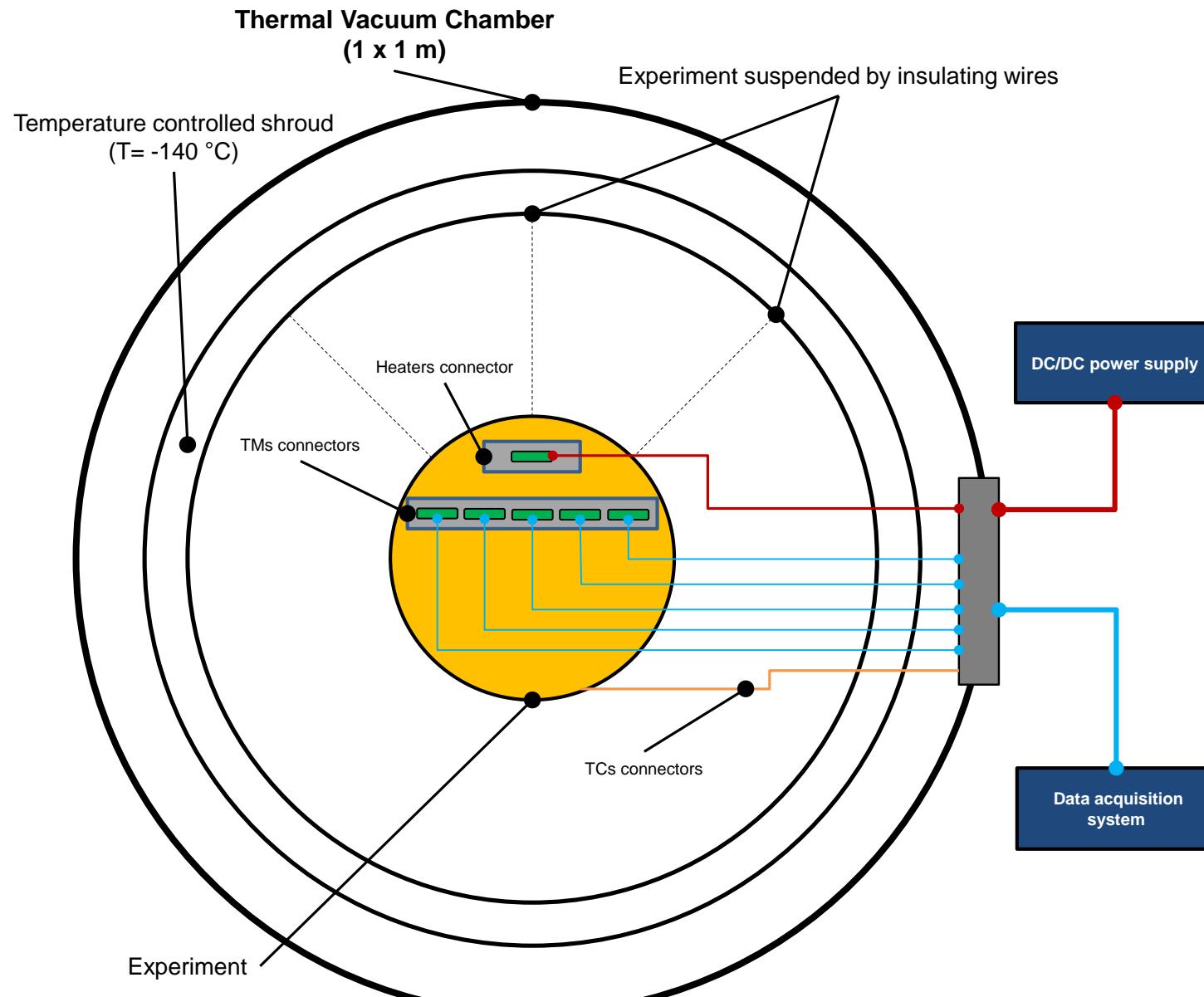
Dummy/heaters/thermistors



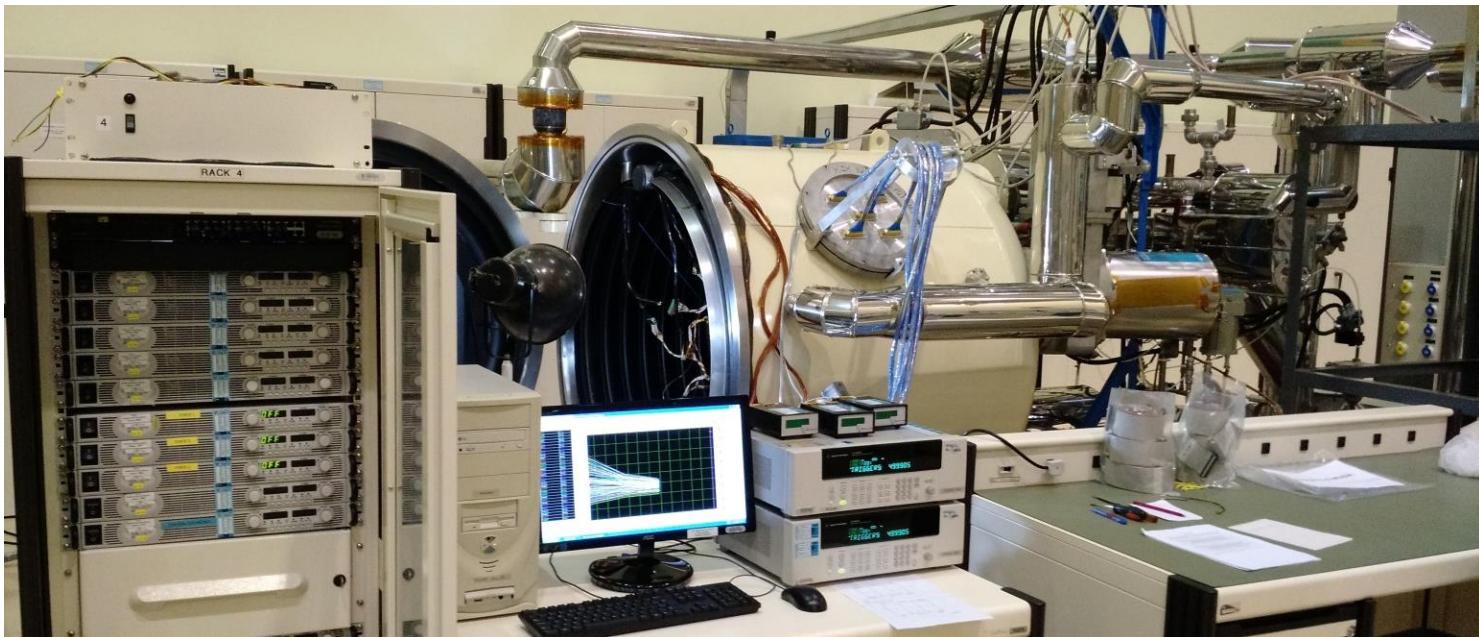
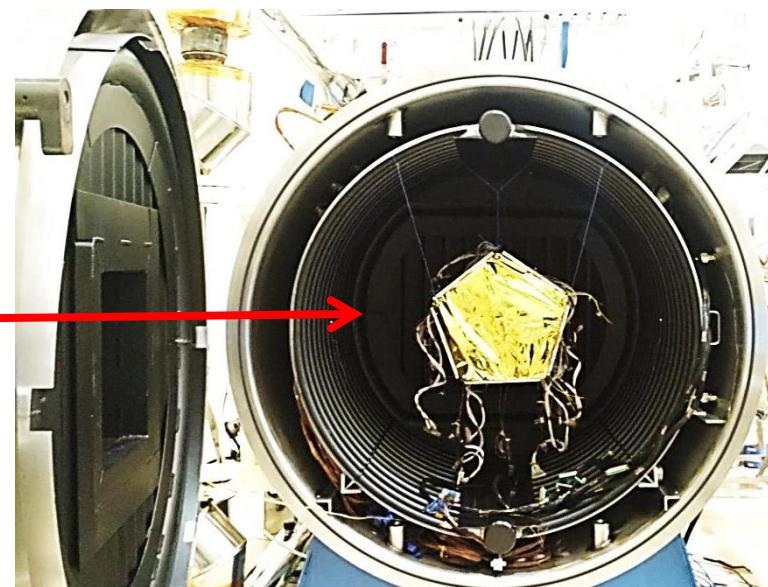
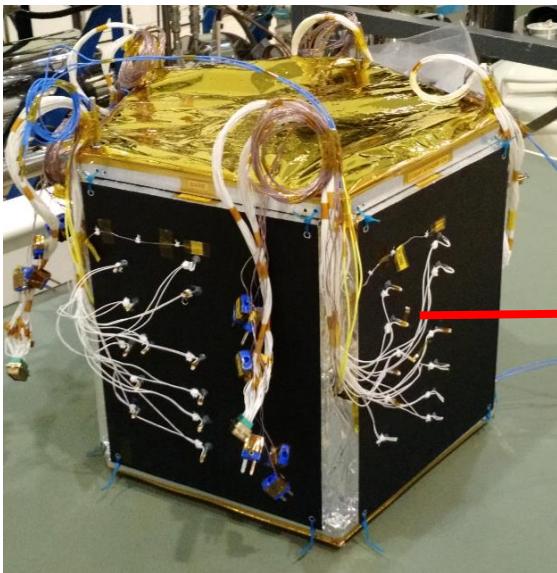
# Apparatus photos



# Test configuration scheme

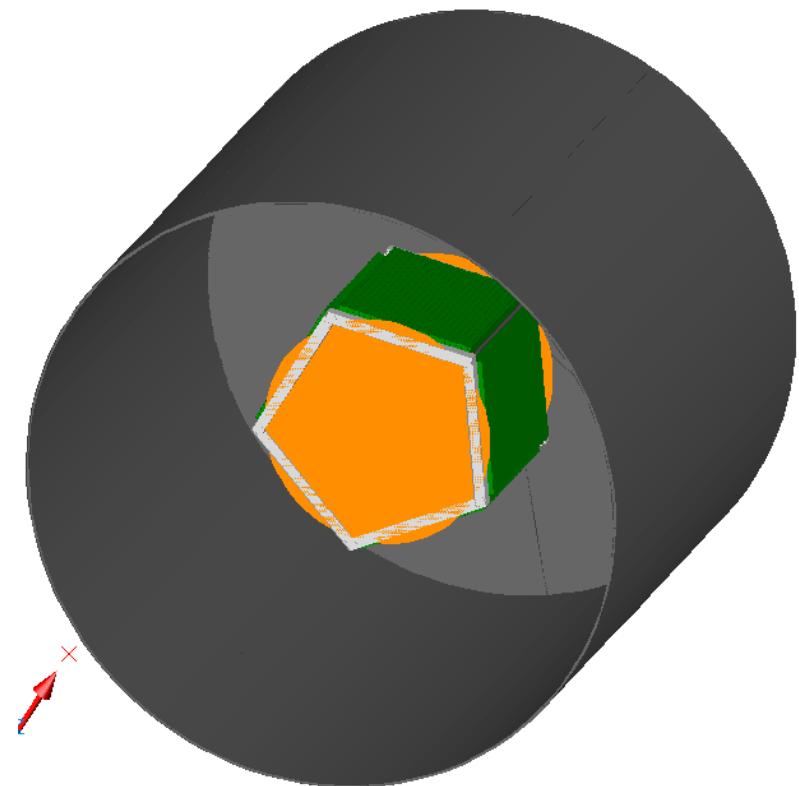
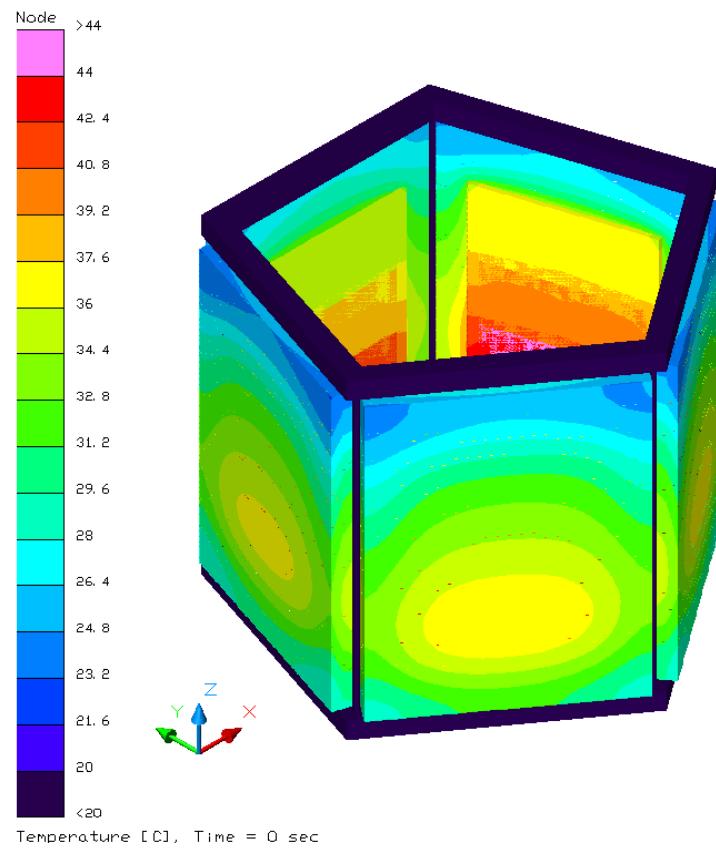


# Test configuration photos

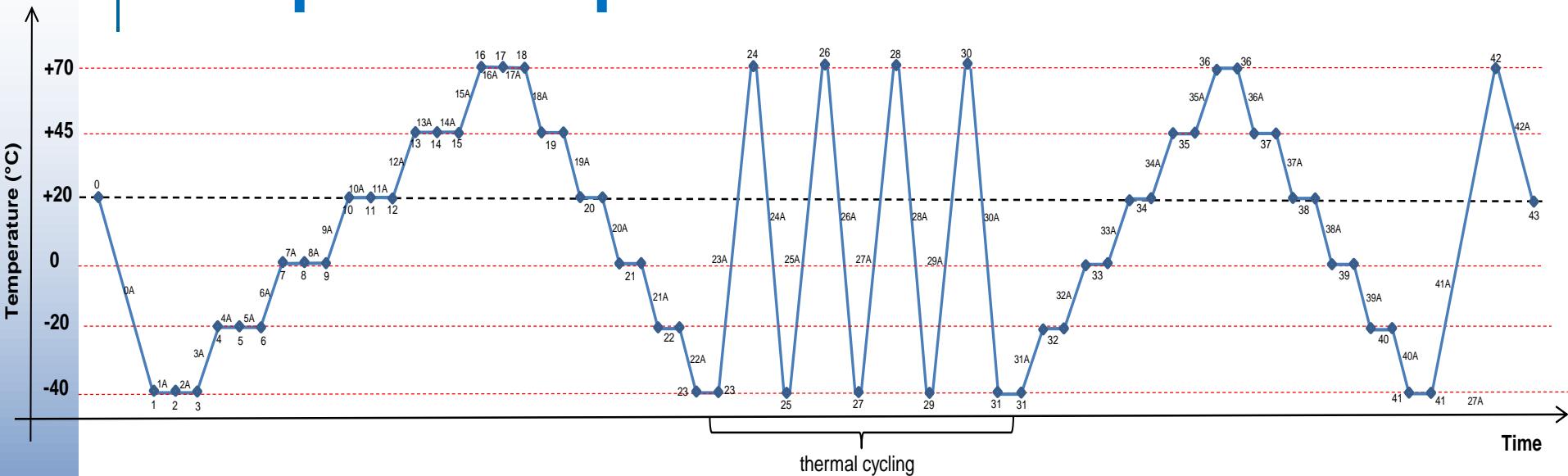


# Test planning

- Thermal Desktop® SINDA/FLUINT simplified TMM
- Heaters heat dissipation estimation
  - Target temperature levels: (-40, -20, 0, +20, +45, +70) °C

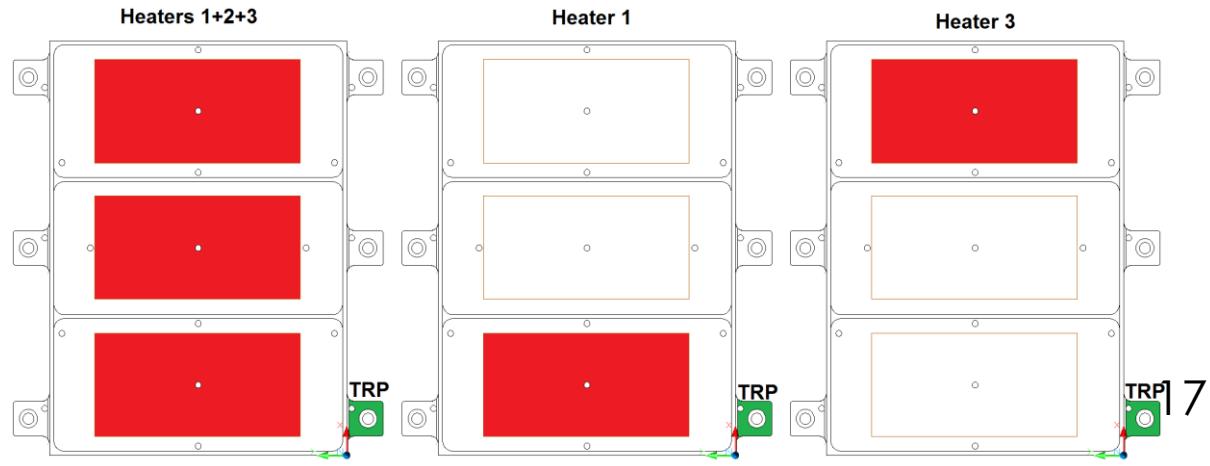


# Test phases - plan



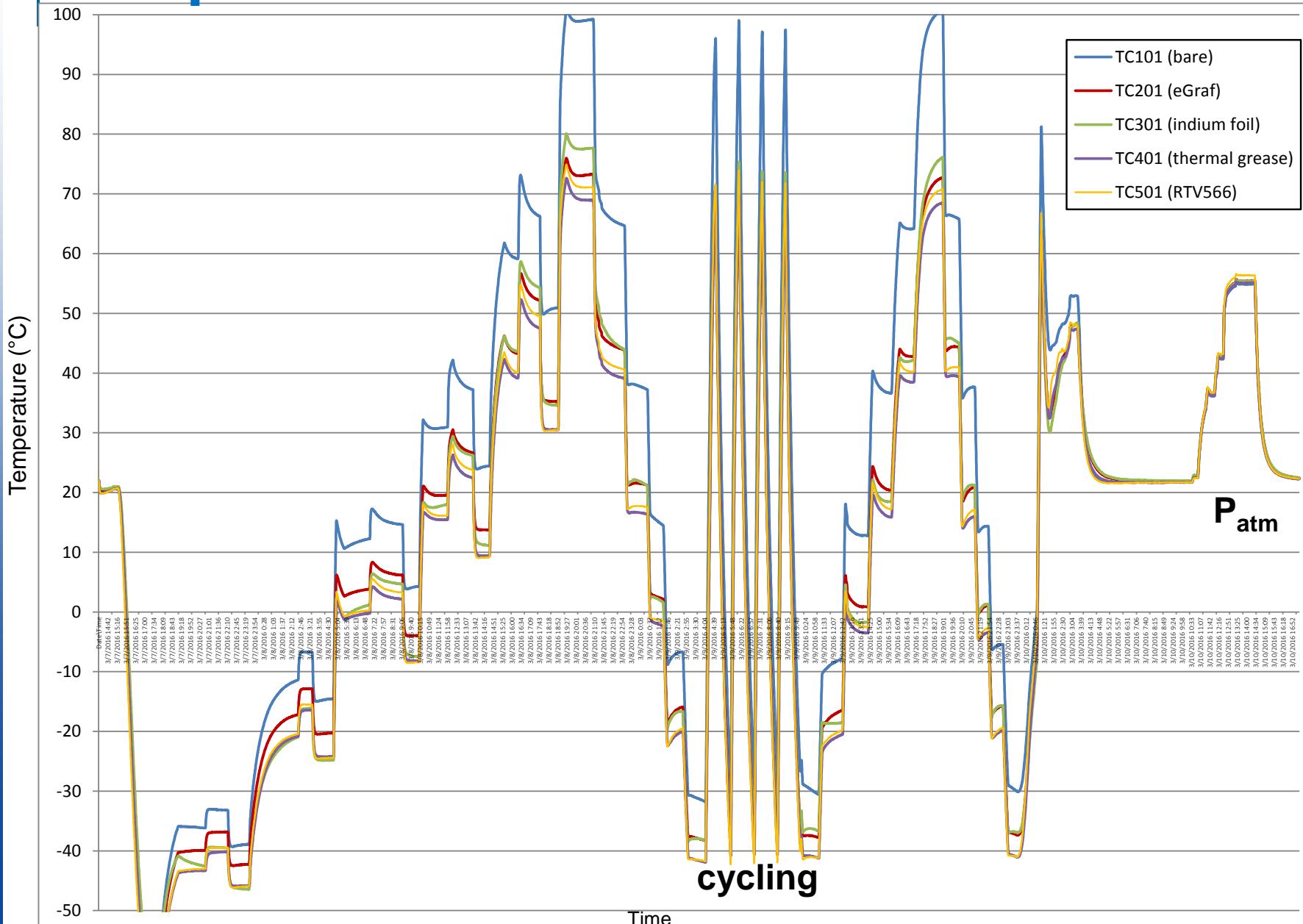
- 6 temperature levels: from -40 to + 70 °C
- Electronic equipment qualification levels (20 °C margin)

- 3 dissipation modes
  - Simulates the non-homogeneous heat dissipation (redundancy or operation modes)



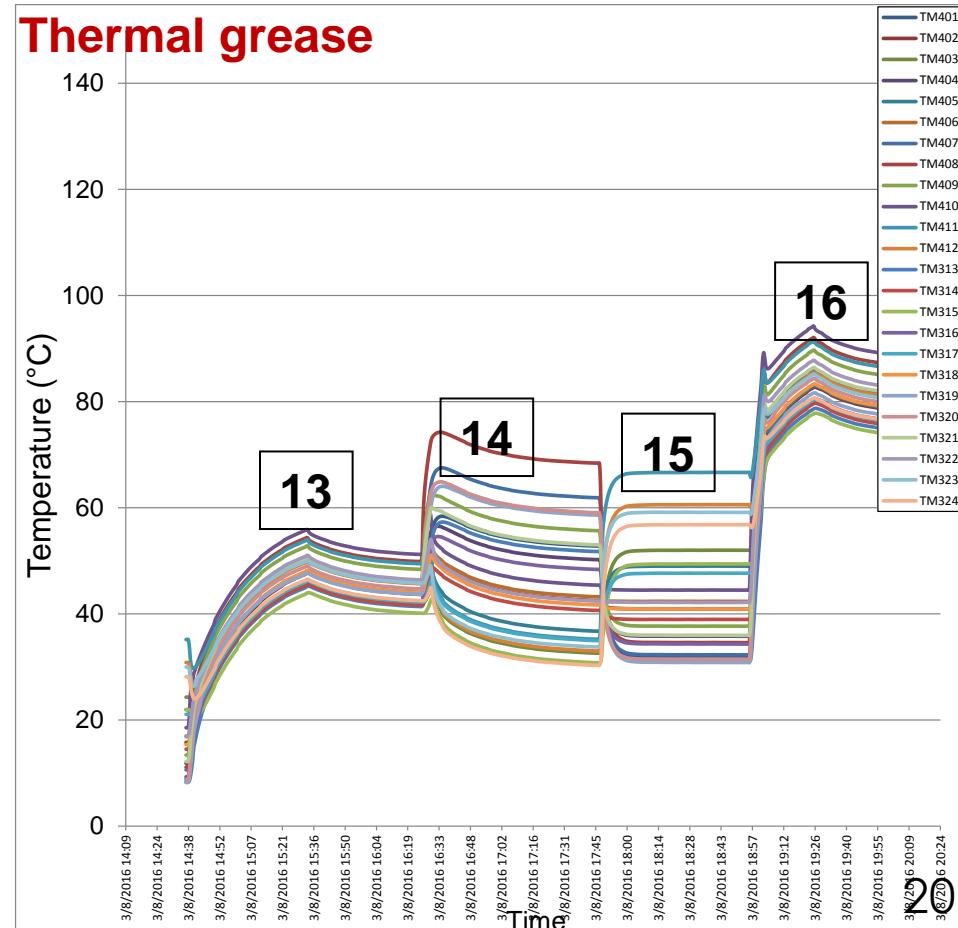
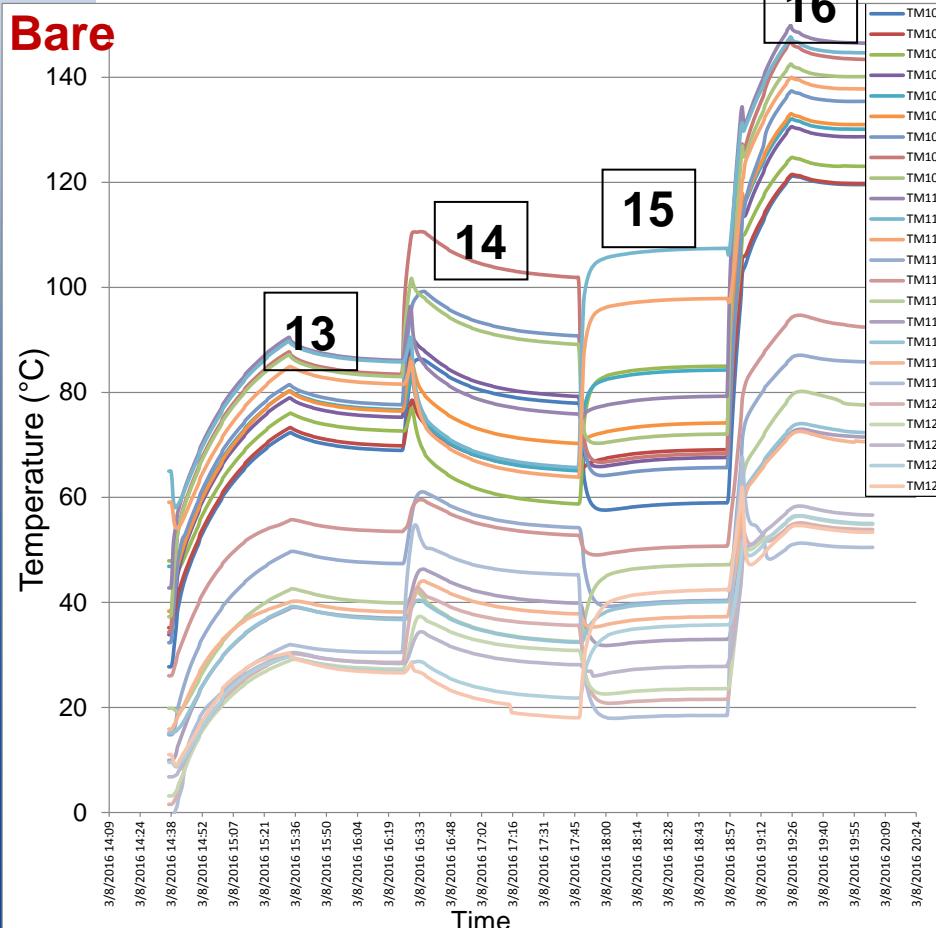
# Preliminary test results

# Test phases - executed



# Bare vs Thermal Grease (vacuum)

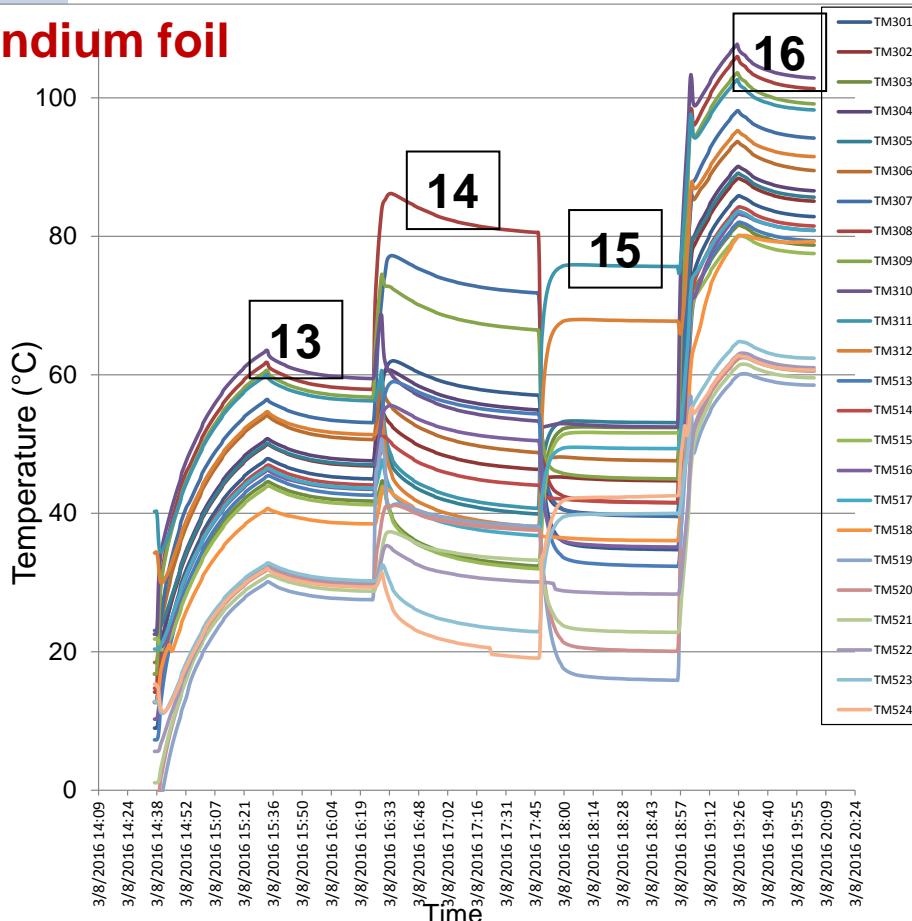
- Phase 13:** H1+H2+H3 @ 33.3W
- Phase 14:** H1 @ 33.3W
- Phase 15:** H3 @ 33.3W
- Phase 16:** H1+H2+H3 @ 47.8W



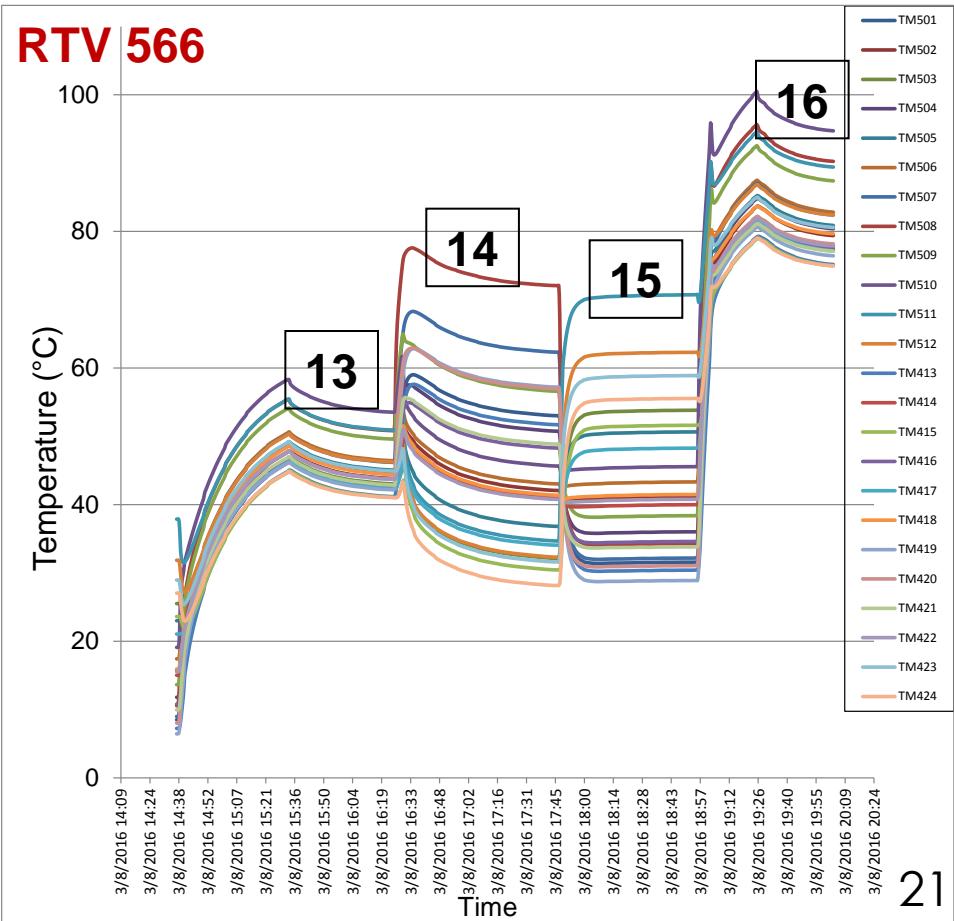
# Indium foil vs RTV 566 (vacuum)

- **Phase 13:** H1+H2+H3 @ 33.3W
- **Phase 14:** H1 @ 33.3W
- **Phase 15:** H3 @ 33.3W
- **Phase 16:** H1+H2+H3 @ 47.8W

Indium foil



RTV 566

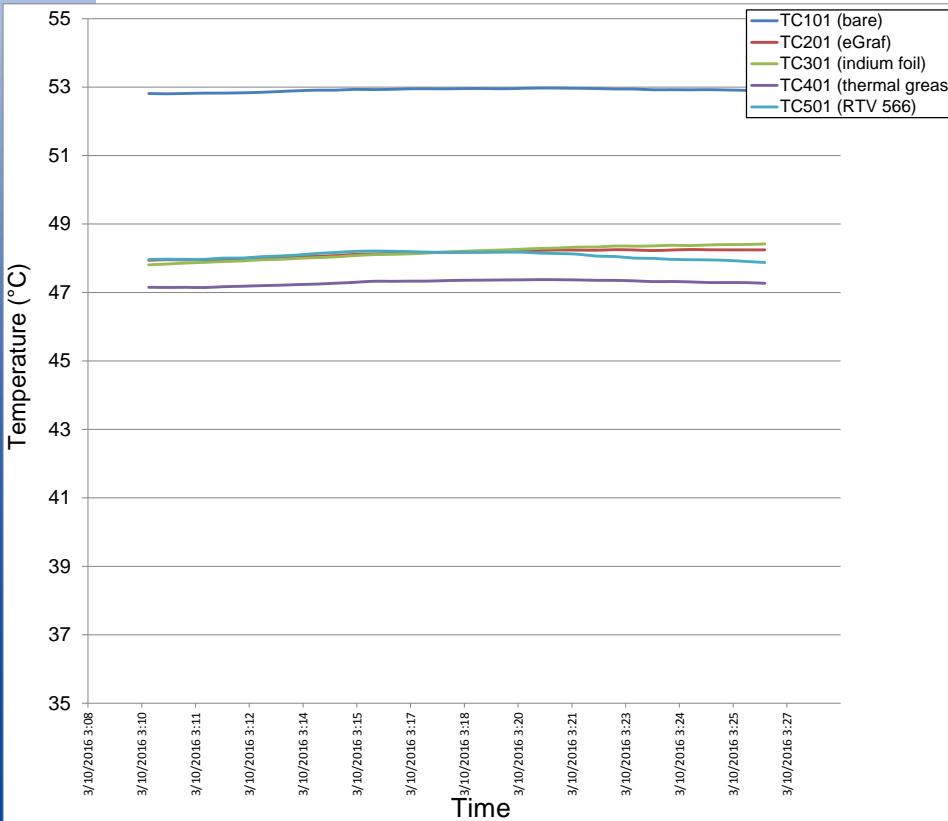


# Vacuum vs P<sub>atm</sub>

**Vacuum**

Q = 9.4 W

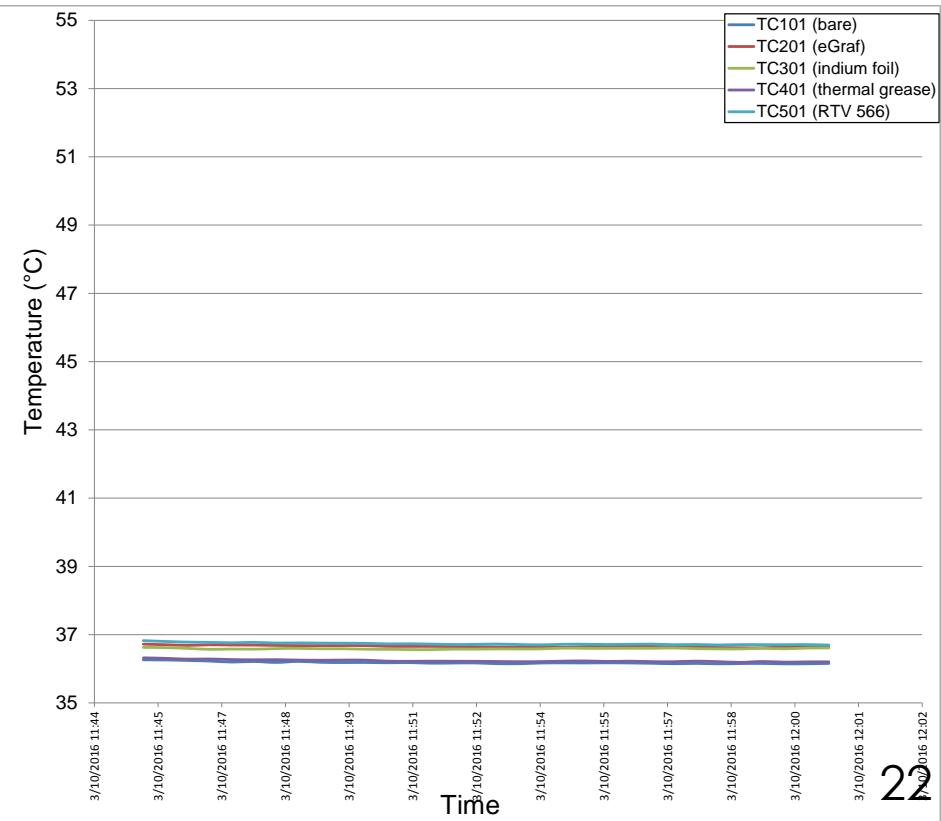
T<sub>shroud</sub> = 25 °C



**P<sub>atm</sub>**

Q = 9.4 W

T<sub>air</sub> = 22 °C

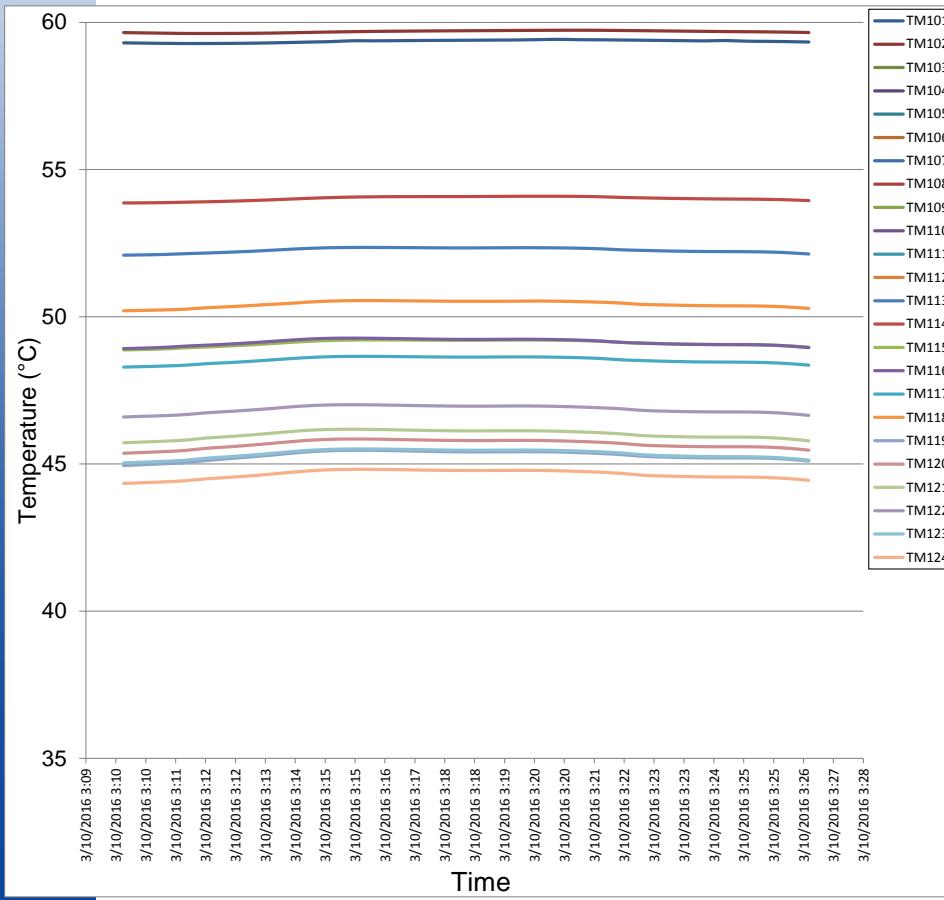


# Vacuum vs $P_{atm}$ – bare interface

**Vacuum**

$Q = 9.4 \text{ W}$

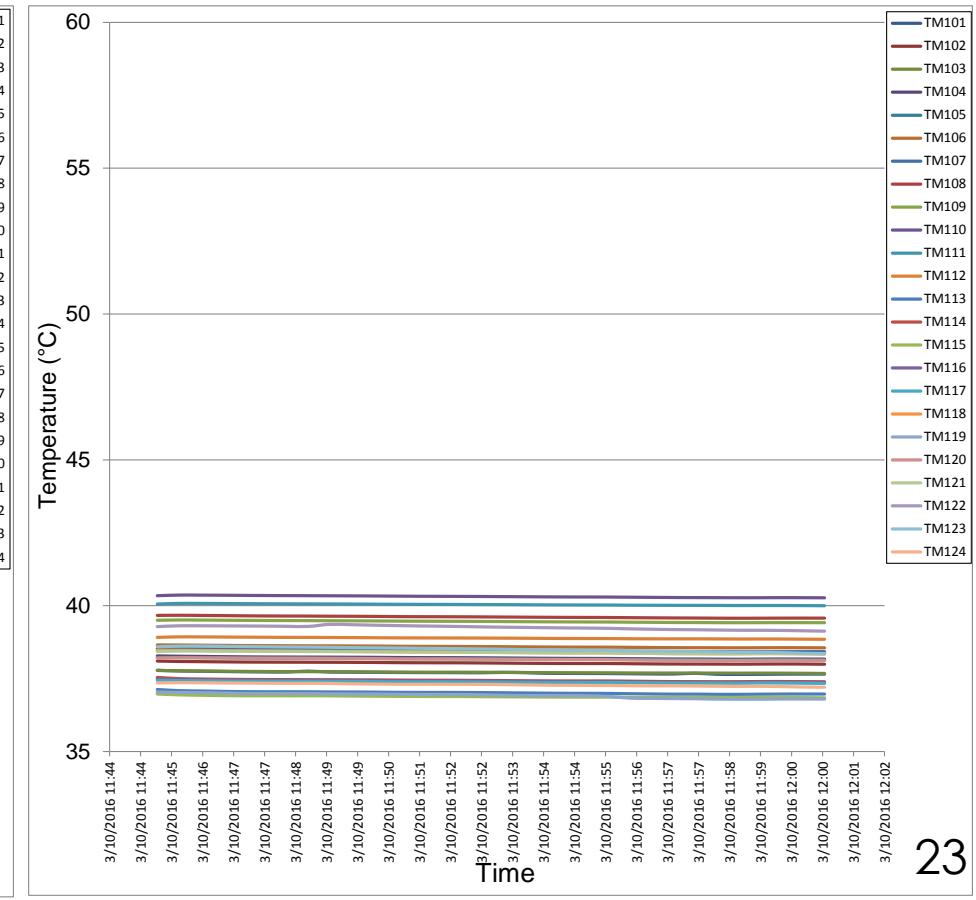
$T_{shroud} = 25 \text{ }^{\circ}\text{C}$



**$P_{atm}$**

$Q = 9.4 \text{ W}$

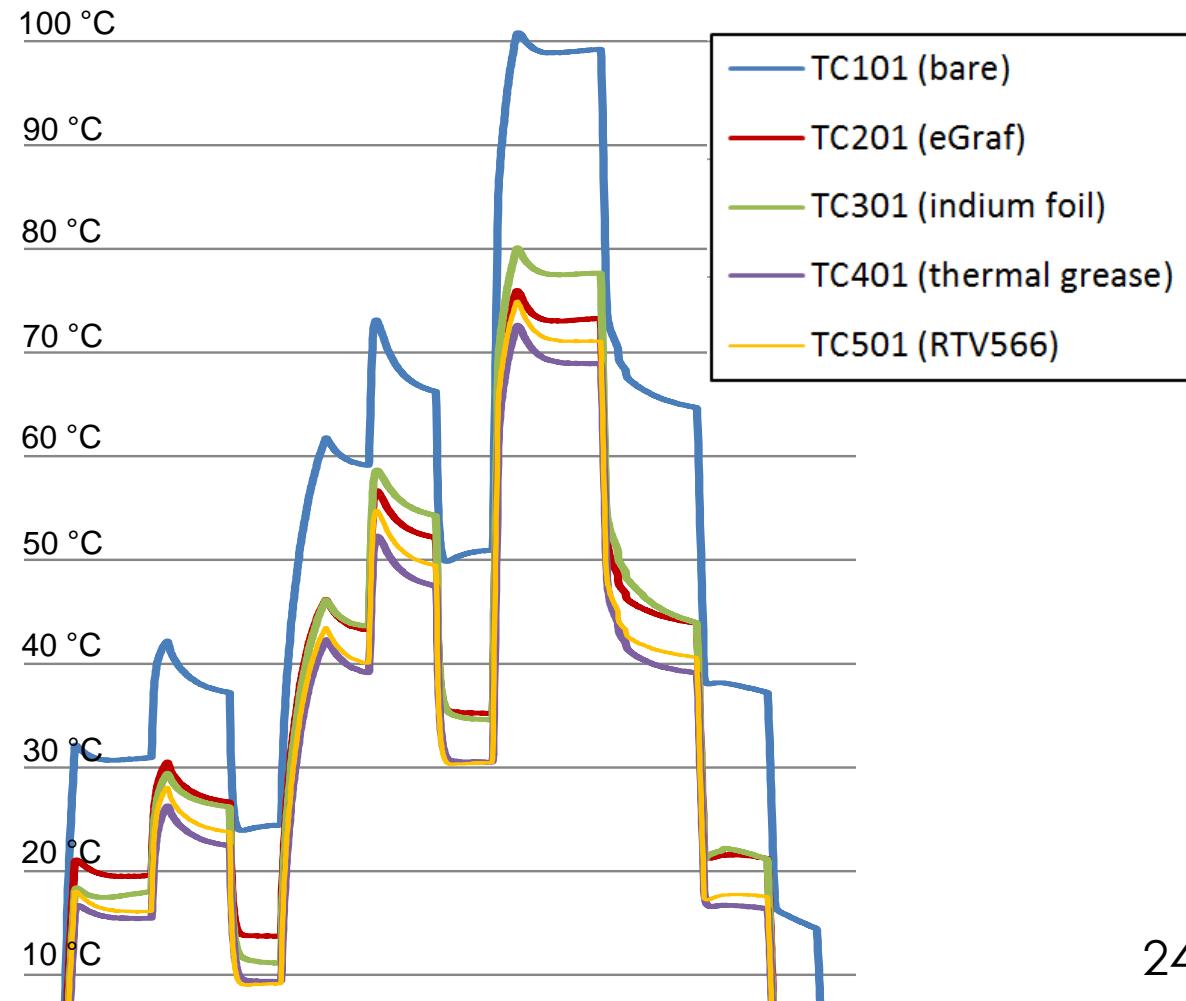
$T_{air} = 22 \text{ }^{\circ}\text{C}$



# Preliminary rank (in vacuum)

Based on the observation of the test temperatures, it can be indicated the following order of thermal performance:

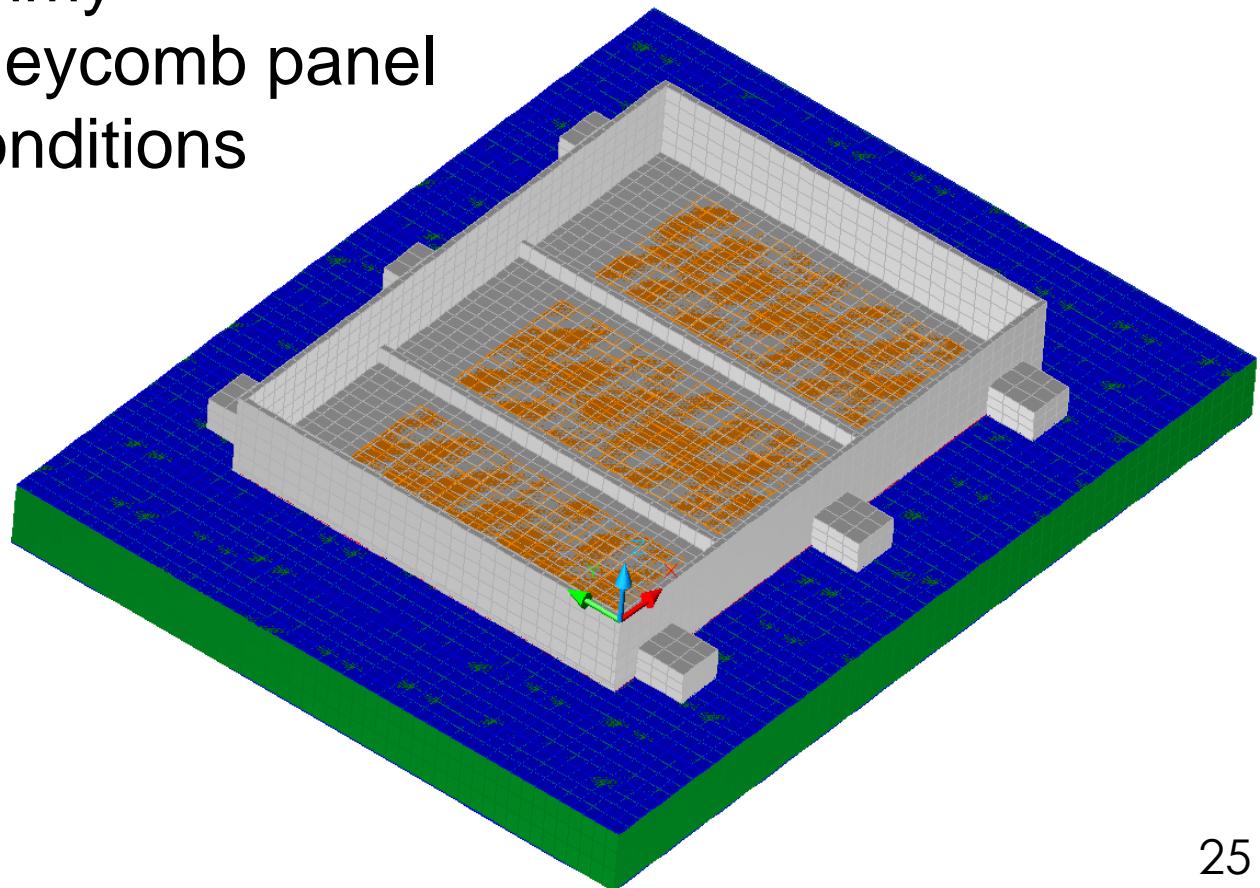
1. Thermal grease
2. RTV 566
3. Indium foil
4. eGraf
5. Bare



## Future work – detailed TMM

- Use the Thermal Desktop® SINDA/FLUINT detailed TMM

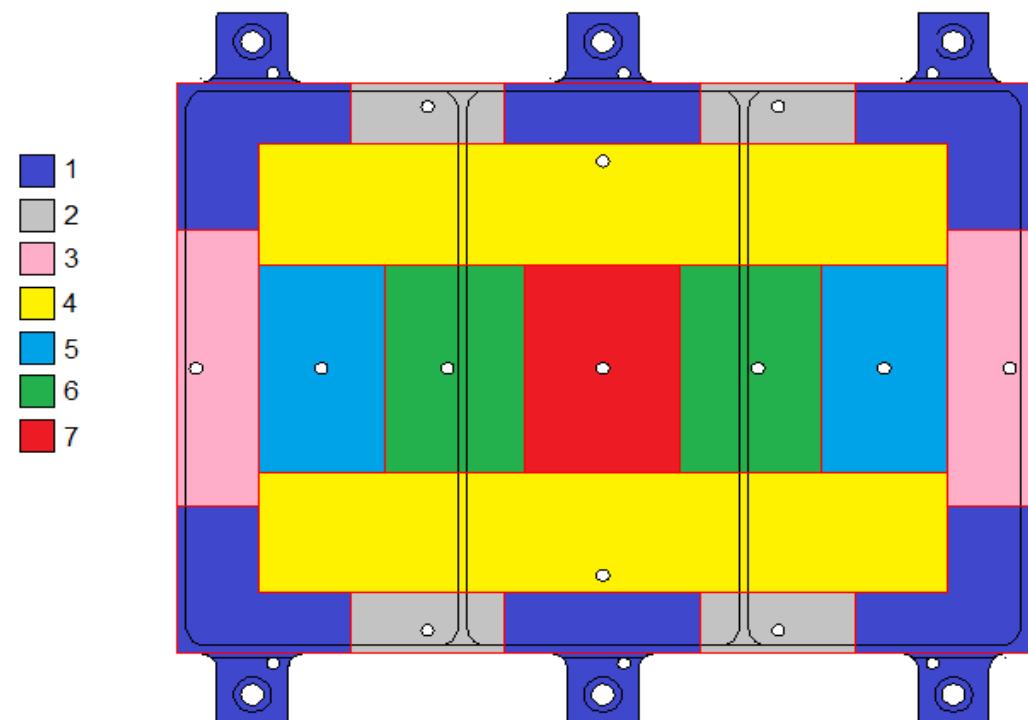
- detailed dummy
- detailed honeycomb panel
- boundary conditions



# Future work – detailed TMM

- Baseplate divided into 7 regions
- Each reagion has a couple of corresponded temperature sensors
- Identification of effective thermal conductance for each reagion – different contact pressure

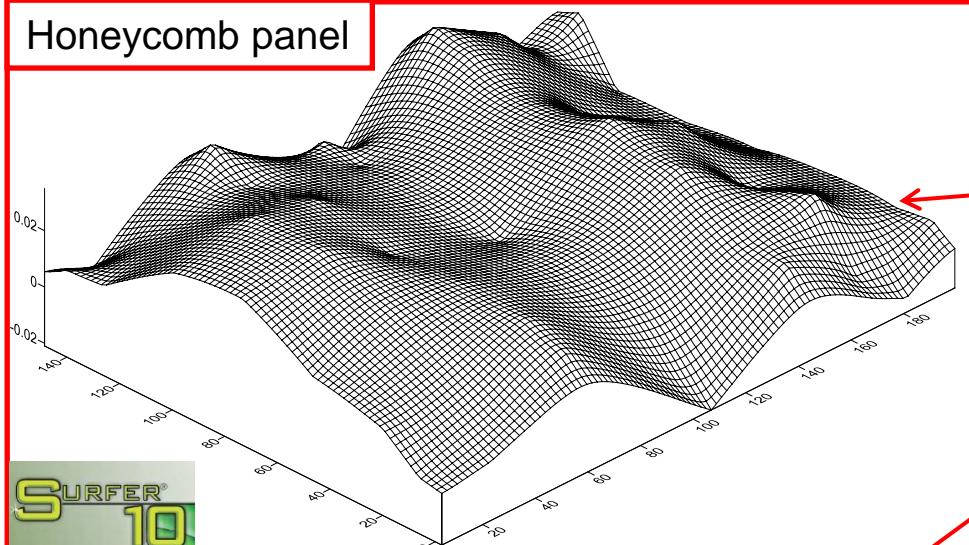
**Proposed regions**



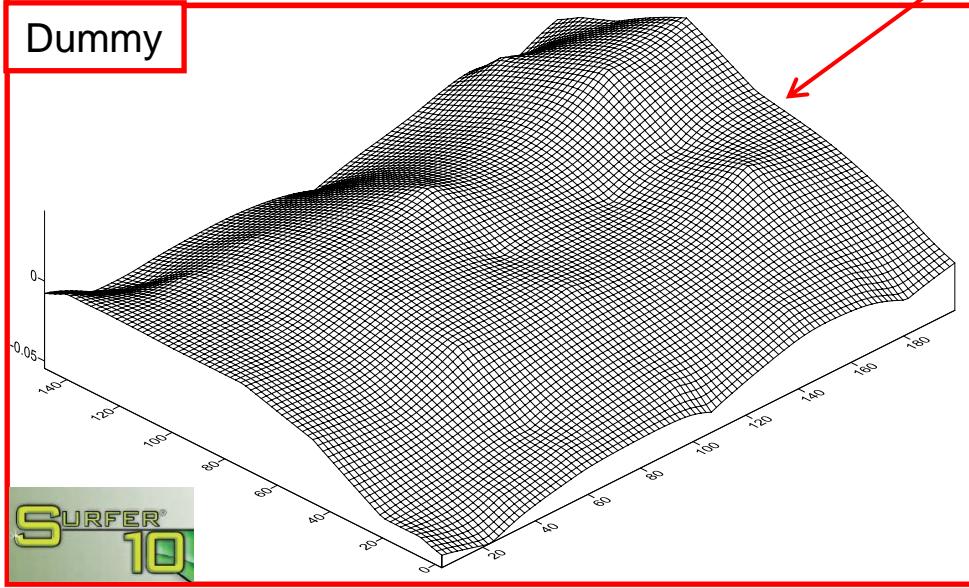
# Influence of flatness/roughness

- All surfaces comply with flatness requirements

Honeycomb panel



Dummy

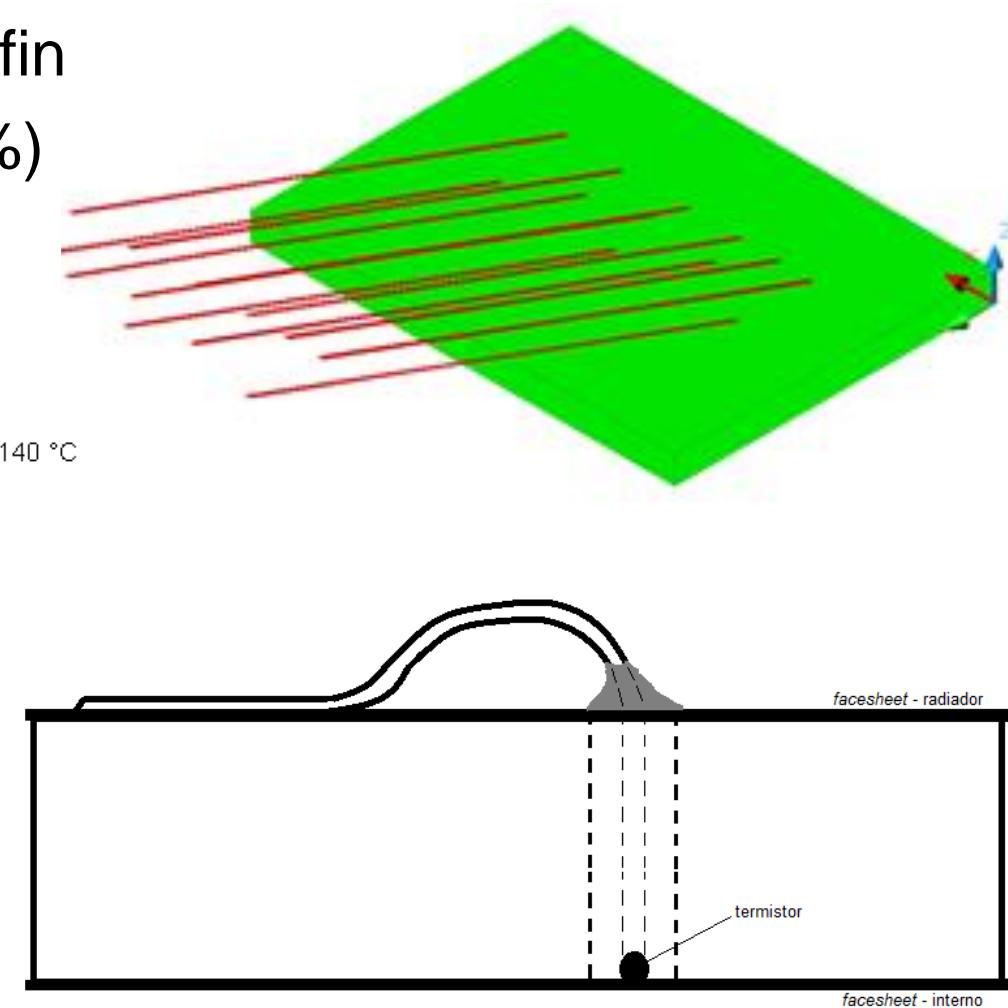
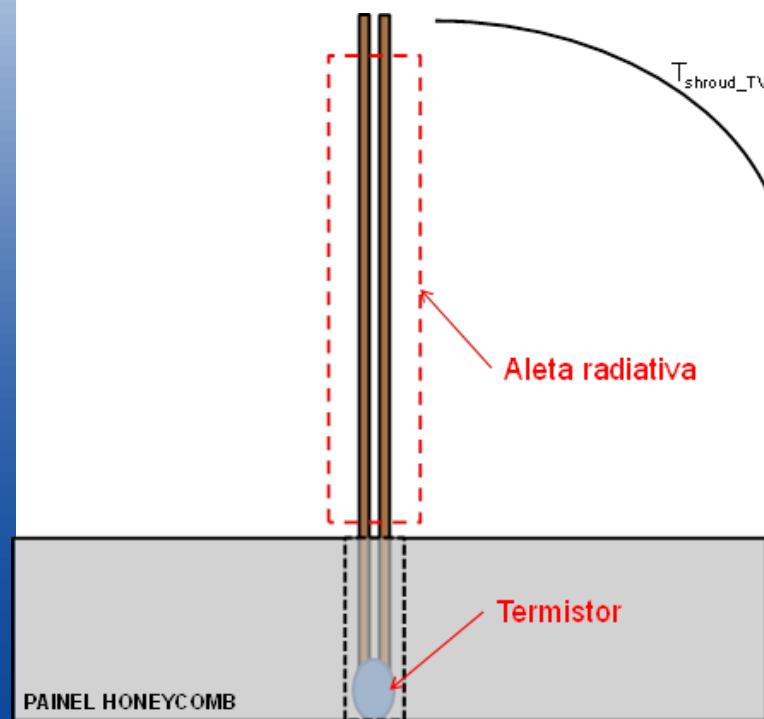


Surface roughness



# Study about thermistor wires influence

- Wires act like radiation fin
- Radiator blockage (~5%)



# Conclusions

- A **unique test setup** has been developed to simulate the interface between satellite electronic equipments and honeycomb panels;
- The test demonstrated that any of the tested materials has a better performance than the bare interface;
- As a preliminary conclusion, the **thermal grease** performed better than the others;
- There are a lot of data to be analysed from the first test;
- It is planned to use the apparatus to test **different instalation methods** for thermal grease and RTV 566 and other materials.